



Reinforced Concrete Cantilever Beam Analysis and Design (ACI 318-14)





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Cantilever beams consist of one span with fixed support at one end and the other end is free. There are numerous typical and practical applications of cantilever beams in buildings, bridges, industrial and special structures.

This example will demonstrate the analysis and design of the rectangular reinforced concrete cantilever beam shown below using ACI 318-14 provisions. Steps of the structural analysis, flexural design, shear design, and deflection checks will be presented. The results of hand calculations are then compared with the reference results and numerical analysis results obtained from the <u>spBeam</u> engineering software program by <u>StructurePoint</u>.





Figure 1 - Rectangular Reinforced Concrete Cantilever Beam



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Code

Building Code Requirements for Structural Concrete (ACI 318-14) and Commentary (ACI 318R-14)

References

spBeam Engineering Software Program Manual v5.00, STRUCTUREPOINT, 2015

Design Data

 $f_{c}' = 4 \text{ ksi normal weight concrete } (w_{c} = 150 \text{ lb/ft}^{3})$ $f_{y} = 60 \text{ ksi}$ Dead load, DL = 12 kip (self-weight is negligible) applied at the free endLive load, LL = 12 kip applied at the free endBeam span length, L = 100 in. = 8.33 ftUse #9 bars for longitudinal reinforcement (A_s = 1.00 in.², d_b = 1.128 in.) Use #4 bars for stirrups (A_s = 0.20 in.², d_b = 0.50 in.) Clear cover = 1.5 in. ACI 318-14 (Table 20.6.1.3.1)





Solution

1. Preliminary Member Sizing

Check the minimum beam depth requirement of <u>ACI 318-14 (Table 9.3.1.1)</u> to waive deflection computations. Using the minimum depth for non-prestressed beams in <u>Table 9.3.1.1</u>.

$$h_{\min} = \frac{l_n}{8} = \frac{100 \text{ in.}}{8} = 12.5 \text{ in.} \text{ (For cantilever beams)}$$
 ACI 318-14 (Table 9.3.1.1)

Therefore, since $h_{min} = 12.5$ in. < h = 24 in. the preliminary beam depth satisfies the minimum depth requirement, and the beam deflection computations are not required.

The width of the rectangular section (b) may be chosen in the following range:

$$\left(\frac{1}{2} \times h = 12 \text{ in.}\right) \le b = 16 \text{ in.} \le \left(\frac{2}{3} \times h = 16 \text{ in.}\right)$$
 o.k.

2. Load and Load combination

For the factored Load

 $w_u = 1.2 \times DL + 1.6 \times LL$

 $P_{u} = 1.2 \times 12 + 1.6 \times 12 = 33.6$ kip

ACI 318-14 (Eq. 5.3.1b)



3. Structural Analysis

Cantilever beams can be analyzed by calculating shear and moment diagrams or using Design Aid tables as shown below:

Shear and Moment Diagrams:







Using Design Aid Tables:

 $V_u = R_A = P_u = 33.6 \text{ kip}$ $M_u = P_u \times L = 33.6 \times 8.33 = 280 \text{ kip-ft}$

CANTILEVER BEAM – CONCENTRATED LOAD AT FREE END



Figure 3 - Design Aid Tables (Beam Design Equations and Diagrams) - PCI Design Handbook

4. Flexural Design

4.1. Required and Provided Reinforcement

For this beam, the moment at the fixed end governs the design as shown in the previous Figure.

 $M_u = 280$ kip-ft

Use #9 bars with 1.5 in. concrete clear cover per <u>ACI 318-14 (Table 20.6.1.3.1)</u>. The distance from extreme compression fiber to the centroid of longitudinal tension reinforcement, *d*, is calculated below:

$$d = h - \left(\text{clear cover} + d_{b, stirrups} + \frac{d_{Longitudinal bar}}{2} \right)$$
$$d = 24 - \left(1.50 + 0.5 + \frac{1.128}{2} \right) = 21.44 \text{ in.}$$

To determine the area of steel, assumptions have to be made whether the section is tension or compression controlled, and regarding the distance between the resultant compression and tension forces along the beam section (*jd*). In this example, tension-controlled section will be assumed so the reduction factor ϕ is equal to 0.9, and *jd* will be taken equal to 0.919d. The assumptions will be verified once the area of steel is finalized.

$$jd = 0.919 \times d = 0.919 \times 21.44 = 19.69$$
 in.

b = 16 in.

The required reinforcement at initial trial is calculated as follows:





$$A_s = \frac{M_u}{\phi \times f_v \times jd} = \frac{280 \times 12,000}{0.9 \times 60,000 \times 19.69} = 3.16 \text{ in.}^2$$

Recalculate 'a' for the actual A_s = 3.16 in.²: $a = \frac{A_s \times f_y}{0.85 \times f'_c \times b} = \frac{3.16 \times 60,000}{0.85 \times 4,000 \times 16} = 3.48$ in.

$$c = \frac{a}{\beta_1} = \frac{3.48}{0.85} = 4.10$$
 in.

Where:

$$\beta_{1} = 0.85 - \frac{0.05 \times (f'_{c} - 4000)}{1000}$$

$$\beta_{1} = 0.85 - \frac{0.05 \times (4000 - 4000)}{1000} = 0.85$$

$$\varepsilon_{t} = \left(\frac{0.003}{c}\right) \times d_{t} - 0.003 = \left(\frac{0.003}{4.10}\right) \times 21.44 - 0.003 = 0.0127 > 0.005$$

Therefore, the assumption that section is tension-controlled is valid.

$$A_{s} = \frac{M_{u}}{\phi \times f_{y} \times \left(d - \frac{a}{2}\right)} = \frac{280 \times 12,000}{0.9 \times 60,000 \times \left(21.44 - \frac{3.48}{2}\right)} = 3.16 \text{ in.}^{2}$$

The minimum reinforcement shall not be less than

$$A_{s,\min} = \frac{3 \times \sqrt{f_c'}}{f_y} \times b_w \times d = \frac{3\sqrt{4000}}{60000} \times 12 \times 21.44 = 1.085 \text{ in.}^2$$
ACI 318-14 (9.6.1.2(a))

And not less than

_

$$A_{s,\min} = \frac{200}{f_y} \times b_w \times d = \frac{200}{60000} \times 12 \times 21.44 = 1.143 \text{ in.}^2$$
ACI 318-14 (9.6.1.2(b))

 $\therefore A_{s,\min} = 1.143 \text{ in.}^2$

$$A_{s,req} = \max \begin{cases} A_s \\ A_{s,\min} \end{cases} = \max \begin{cases} 3.16 \\ 1.143 \end{cases} = 3.16 \text{ in.}^2$$

Provide 4 – #9 bars:

$$A_{s, prov} = 4 \times 1.00 = 4.00 \text{ in.}^2 > A_{s, req} = 3.16 \text{ in.}^2$$





spBeam Manual (Eq. 2-96)

spBeam Manual (Figure 2.21)

4.2. Spacing of Longitudinal Reinforcement

$$w_{bend} = \left(1 - \frac{\sqrt{2}}{2}\right) + \left(r - \frac{d_{b,longitudinal}}{2}\right)$$

Where r is the inside radius of bend for stirrup = 4 x stirrup radius = 4 x 0.50/2 = 1 in.

$$d_s = \text{Side Cover} + d_{b,stirrup} + w_{bend} + \frac{d_{b,longitudinal}}{2}$$

$$d_s = 1.50 + 0.50 + 0.13 + \frac{1.128}{2} = 2.69$$
 in.

$$s_{provided} = \frac{(b-2 \times d_s)}{\#of \ bars - 1} = \frac{(16-2 \times 2.69)}{4-1} = 3.539$$
 in.

Where:



Figure 4 – Width Due to Stirrup Bend (spBeam Manual – Figure 2.21)

The maximum allowed spacing (s_{max}):

$$s_{\max} = 15 \left(\frac{40000}{f_s}\right) - 2.5c_c \le 12 \left(\frac{40000}{f_s}\right)$$
 ACI 318-14 (Table 24.3.2)

 c_c = the least distance from surface of reinforcement to the tension face = 2.0 in.

Use
$$f_s = \frac{2}{3} f_y = 40000 \text{ psi}$$

 $s_{\text{max}} = \min \begin{cases} 15 \times \left(\frac{40000}{40000}\right) - 2.5 \times 2.0 \\ 12 \times \left(\frac{40000}{40000}\right) \end{cases} = \min \begin{cases} 10 \\ 12 \end{cases} = 10 \text{ in.}$

The minimum allowed spacing (s_{min}):

$$s_{\min} = d_b + \max \begin{cases} 1 \\ d_b \\ 1.33 \times \max.agg. \end{cases}$$
CRSI 2002 (Figure 12-9)





Where the maximum aggregate size is ³/₄"

 $s_{\min} = 1.00 + \max \left\{ \begin{array}{c} 1.00 \\ 1.128 \\ 1.33 \times 0.75 = 1.00 \end{array} \right\} = 1.00 + 1.128 = 2.256 \text{ in.}$

 $s_{min} = 2.256 \mbox{ in.} < s_{provided} = 3.539 \mbox{ in.} < s_{max} = 10.000 \mbox{ in.}$

Therefore, 4 - #9 bars are <u>o.k.</u>



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5. Shear Design

Structure P



Figure 5 – Shear Diagram for Cantilever Beam

$$V_{\mu} = V_{\mu @ d} = 33.6$$
 kips

Shear strength provided by concrete

$$\phi V_c = \phi \times 2 \times \sqrt{f_c} \times b_w \times d$$

$$\phi V_c = 0.75 \times 2 \times \sqrt{4000} \times 16 \times 21.44 = 32.54 \text{ kips}$$

$$\frac{\phi V_c}{2} = 16.27 \text{ kips} < V_u = 33.6 \text{ kips}$$
Since $V_c \ge \phi V_c$ show reinforcement is required

Since $V_u > \phi V_c/2$, shear reinforcement is required.

Try # 4, Grade 60 two-leg stirrups ($A_v = 2 \ge 0.40 \text{ in.}^2$).

The nominal shear strength required to be provided by shear reinforcement is

$$V_s = V_n - V_c = \frac{V_{u@d}}{\phi} - \frac{\phi V_c}{\phi} = \frac{33.6}{0.75} - \frac{32.54}{0.75} = 1.42$$
 kips

If V_s is greater than $8\sqrt{f'_c}b_w d$, then the cross-section has to be revised as <u>ACI 318-14</u> limits the shear capacity to be provided by stirrups to $8\sqrt{f'_c}b_w d$. $8 \times \sqrt{f'_c} \times b_w \times d = 8 \times \sqrt{4000} \times 16 \times 21.44 = 173.53$ kips $\rightarrow \therefore$ section is adequate





$$\left(\frac{A_{v}}{s}\right)_{req} = \frac{V_{u@d} - \phi V_{c}}{\phi \times f_{yt} \times d} = \frac{33.60 - 32.54}{0.75 \times 60 \times 21.44} = 0.0011 \text{ in.}^{2}/\text{in.} \qquad \underline{ACI 318-14 (22.5.10.5.3)}$$

$$\left(\frac{A_{v}}{s}\right)_{min} = \max\left\{\frac{\frac{0.75 \times \sqrt{f_{c}} \times b_{w}}{f_{yt}}}{50 \times b_{w}}\right\}$$

$$\left(\frac{A_{v}}{s}\right)_{min} = \max\left\{\frac{\frac{0.75 \times \sqrt{4000} \times 16}{60000}}{50 \times 16}\right\} = \max\left\{\frac{0.0126}{0.0133}\right\} = 0.0133 \text{ in.}^{2}/\text{in.} > \left(\frac{A_{v}}{s}\right)_{req} = 0.0011 \text{ in.}^{2}/\text{ in.}$$

$$\therefore \left(\frac{A_{v}}{s}\right)_{req} = 0.0133 \text{ in.}^{2}/\text{ in.} \quad (\text{Minimum transverse reinforcement governs})$$

$$s_{max} = \frac{A_{v}}{2} = \frac{0.40}{2} = 30 \text{ in.}$$

$$s_{req} = \frac{A_v}{\left(\frac{A_v}{s}\right)_{req}} = \frac{0.40}{0.0133} = 30 \text{ i}$$

Check whether the required spacing based on the shear demand meets the spacing limits for shear reinforcement per <u>ACI 318-14 (9.7.6.2.2)</u>.

$$4 \times \sqrt{f_c} \times b_w \times d = 4 \times \sqrt{4000} \times 16 \times 21.44 = 86.77 \text{ kips} > V_s = 1.42 \text{ kips}$$

Therefore, maximum stirrup spacing shall be the smallest of *d*/2 and 24 *in*. <u>ACI 318-14 (Table 9.7.6.2.2)</u>

$$s_{\max} = \min \left\{ \frac{d/2}{24 \text{ in.}} \right\} = \min \left\{ \frac{21.44/2}{24 \text{ in.}} \right\} = \min \left\{ \frac{10.72 \text{ in.}}{24 \text{ in.}} \right\} = 10.72 \text{ in.}$$

This value governs over the required stirrup spacing of 30 in which was based on the demand.

Therefore, *s_{max}* value is governed by the spacing limit per <u>ACI 318-14 (9.7.6.2.2)</u>, and is equal to 10.72 in.

$$s_{provided} = \frac{L - 2 \times (\text{Location of First Stirrup})}{\# \text{ of Stirrups} - 1} = \frac{100 \text{ in.} - 2 \times (3 \text{ in.})}{10 - 1} = 10.444 \text{ in.} < s_{\text{max}} = 10.720 \text{ in.}$$

Use 10 - # 4 @ 10.444 in. stirrups (it is more practical to round the provided spacing to 10 in., the provided spacing is kept as 10.444 in. for comparison reasons with <u>spBeam</u> results).

$$\phi V_n = \frac{\phi \times A_v \times f_{yt} \times d}{s} + \phi V_c$$

$$\frac{ACI 318-14 (22.5.1.1 \text{ and } 22.5.10.5.3)}{6V_n}$$

$$\phi V_n = \frac{0.75 \times 0.40 \times 60 \times 21.44}{10.444} + 32.54 = 36.94 + 32.54 = 69.48 \text{ kips} > V_{u@d} = 33.60 \text{ kips}$$

$$o.k.$$

Use 10 - # 4 @ 10.444 in. o.c., Place 1st stirrup 3 in. from the face of the column.

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6. Deflection Control (Serviceability Requirements)

Since the preliminary beam depth met minimum depth requirement, the deflection calculations are not required. However, the calculations of immediate and time-dependent deflections are covered in detail in this section for illustration and comparison with <u>spBeam</u> model results for cantilever beam.

6.1. Immediate (Instantaneous) Deflections

Elastic analysis for three service load levels $(D, D + L_{sustained}, D + L_{Full})$ is used to obtain immediate deflections of the cantilever beam in this example. However, other procedures may be used if they result in predictions of deflection in reasonable agreement with the results of comprehensive tests.

The effective moment of inertia procedure described in the Code is considered sufficiently accurate to estimate deflections. The effective moment of inertia, I_e , was developed to provide a transition between the upper and lower bounds of I_g and I_{cr} as a function of the ratio M_{cr}/M_a .

Unless deflections are determined by a more comprehensive analysis, immediate deflection shall be computed using elastic deflection equations using I_e from Eq. (24.2.3.5a) at midspan for simple and continuous spans, and <u>at the support for cantilevers</u>. <u>ACI 318-14 (24.2.3.7)</u>

The effective moment of inertia (I_e) is used to account for the cracking effect on the flexural stiffness of the beam. I_e for uncracked section ($M_{cr} > M_a$) is equal to I_g . When the section is cracked ($M_{cr} < M_a$), then the following equation should be used:

$$I_e = \left(\frac{M_{cr}}{M_a}\right)^3 I_g + \left[1 - \left(\frac{M_{cr}}{M_a}\right)^3\right] I_{cr} \le I_g \qquad \underline{ACI 318-14 (Eq. 24.2.3.5a)}$$

Where:

 M_a = Maximum moment in member due to service loads at stage deflection is calculated.

The values of the maximum moments for the three service load levels are calculated from structural analysis as shown previously (sustained live load = 0).

$$M_{DL} = M_{DL+LL_sustained} = P_{DL} \times L = 12 \times 8.33 = 100$$
 kip-ft

$$M_{DL+LL} = (P_{DL} + P_{LL}) \times L = (12 + 12) \times 8.33 = 200$$
 kip-ft

 M_{cr} = cracking moment.

$$M_{cr} = \frac{f_r I_g}{Y_r} = \frac{(474.34) \times (18432)}{12} \times \frac{1}{12000} = 60.72 \text{ kip-ft}$$

ACI 318-14 (Eq. 24.2.3.5b)

 f_r = Modulus of rapture of concrete.

$$f_r = 7.5\lambda\sqrt{f'_c} = 7.5 \times 1.0 \times \sqrt{4000} = 474.34 \text{ psi}$$
 ACI 318-14 (Eq. 19.2.3.1)



 I_g = Moment of inertia of the gross uncracked concrete section

$$I_g = \frac{b \times h^3}{12} = \frac{16 \times 24^3}{12} = 18432 \text{ in.}^4$$
$$y_t = \frac{h}{2} = \frac{24}{2} = 12 \text{ in.}$$

 I_{cr} = moment of inertia of the cracked section transformed to concrete.

CAC Concrete Design Handbook 4th Edition (5.2.3)

The critical section at midspan is reinforced with 4 - #9 bars.



Figure 6 - Gross and Cracked Moment of Inertia of Rectangular Section (PCA Notes Table 10-2)

 E_c = Modulus of elasticity of concrete.

$$E_{c} = w_{c}^{1.5} 33\sqrt{f_{c}^{7}} = 150^{1.5} \times 33 \times \sqrt{4000} = 3834.3 \text{ ksi}$$

$$ACI 318-14 (19.2.2.1.a)$$

$$n = \frac{E_{s}}{E_{c}} = \frac{29000}{3834.3} = 7.56$$

$$PCA Notes on ACI 318-11 (Table 10-2)$$

$$B = \frac{b}{nA_{s}} = \frac{16}{7.56 \times (4 \times 1.00)} = 0.529 \text{ in.}^{-1}$$

$$PCA Notes on ACI 318-11 (Table 10-2)$$

$$kd = \frac{\sqrt{2dB+1}-1}{B} = \frac{\sqrt{2 \times 21.44 \times 0.529+1}-1}{0.529} = 7.31 \text{ in.}$$

$$PCA Notes on ACI 318-11 (Table 10-2)$$

$$I_{cr} = \frac{b(kd)^{3}}{3} + nA_{s}(d-kd)^{2}$$

$$PCA Notes on ACI 318-11 (Table 10-2)$$

$$I_{cr} = \frac{16 \times 7.31^{3}}{3} + 7.56 \times (4 \times 1.00) \times (21.44 - 7.31)^{2} = 8120 \text{ in.}^{4}$$
For dead load - service load level:





$$I_{ec} = I_{cr} + \left(I_g - I_{cr}\right) \left(\frac{M_{cr}}{M_a}\right)^3 \text{, since } M_{cr} = 60.72 \text{ kip-ft} < M_a = 100.00 \text{ kip-ft} \qquad \underline{ACI 318-14 (24.2.3.5a)}$$

$$I_e = 8120 + (18432 - 8120) \left(\frac{60.72}{100.00}\right)^3 = 10428 \text{ in.}^4$$

The following Table provides a summary of the required parameters and calculated values needed for deflection calculation.

Table 1 – Effective Moment of Inertia Calculations (at midspan)								
т		M _a , kip-ft		М	I _e , in. ⁴			
in. ⁴	n.4 $in.4$	D	D +	D +	kip-ft	D	D +	D +
		D	LL _{Sus}	L_{full}	1	D	LL _{Sus}	L_{full}
18432	8120	100.00	100.00	200.00	60.72	10428	10428	8409

After obtaining the effective moment of inertia, the maximum span deflection for the cantilever beam (at the free end) can be obtained from any available procedures or design aids (see Figure 3).

$$\Delta_{\max} = \frac{1}{3} \times \frac{P \times L^3}{E_c \times I_e} \text{ (at the free end)}$$
$$\Delta_{DL} = \frac{1}{3} \times \frac{12 \times 100^3}{(3834.25 \times 10^3) \times 10428} = 0.100 \text{ in.}$$
$$\Delta_{Total} = \frac{1}{3} \times \frac{(12+12) \times 100^4}{(3834.25 \times 10^3) \times 8409} = 0.248 \text{ in.}$$

 $\Delta_{LL} = \Delta_{Total} - \Delta_{DL} = 0.248 - 0.100 = 0.148 \text{ in.} < \frac{L}{360} = \frac{100}{360} = 0.278 \text{ in.} \quad (o.k.) \qquad \underline{ACI 318-14 (Table 24.2.2)}$



6.2. Time-Dependent (Long-Term) Deflections (Δlt)

The additional time-dependent (long-term) deflection resulting from creep and shrinkage (Δ_{cs}) are estimated as follows.

$$\Delta_{cs} = \lambda_{\Lambda} \times (\Delta_{sust})_{Inst}$$
PCA Notes on ACI 318-11 (9.5.2.5 Eq. 4)

The total time-dependent (long-term) deflection is calculated as:

$$\left(\Delta_{total}\right)_{lt} = \left(\Delta_{sust}\right)_{Inst} \times \left(1 + \lambda_{\Delta}\right) + \left(\left(\Delta_{total}\right)_{Inst} - \left(\Delta_{sust}\right)_{Inst}\right)$$

$$\underline{CSA \ A23.3-04 \ (N9.8.2.5)}$$

Where:

 $(\Delta_{sust})_{Inst}$ = Immediate (instantaneous) deflection due to sustained load, in.

$$\lambda_{\Delta} = \frac{\xi}{1+50\rho'} \frac{ACI 318-14 (24.2.4.1.1)}{1+50\rho'}$$

 $(\Delta_{total})_{lt}$ = Time-dependent (long-term) total deflection, in.

 $(\Delta_{total})_{Inst}$ = Total immediate (instantaneous) deflection, in.

 $\xi = 2$, consider the sustained load duration to be 60 months or more. <u>ACI 318-14 (Table 24.2.4.1.3)</u>

 $\rho' = 0$, conservatively.

$$\lambda_{\Delta} = \frac{2}{1+50\times0} = 2$$

$$\Delta_{cs} = 2 \times 0.100 = 0.200$$
 in.

 $\Delta_{cs} + \Delta_{LL} = 0.200 + 0.148 = 0.348 \text{ in.} < \frac{L}{240} = \frac{100}{240} = 0.417 \text{ in.} \quad (o.k.) \qquad \underline{ACI 318-14 (Table 24.2.2)}$

 $(\Delta_{total})_{tt} = 0.100 \times (1+2) + (0.248 - 0.100) = 0.448$ in.

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7. Cantilever Beam Analysis and Design - spBeam Software

<u>spBeam</u> is widely used for analysis, design and investigation of beams, and one-way slab systems (including standard and wide module joist systems) per latest American (ACI 318-14) and Canadian (CSA A23.3-14) codes. <u>spBeam</u> can be used for new designs or investigation of existing structural members subjected to flexure, shear, and torsion loads. With capacity to integrate up to 20 spans and two cantilevers of wide variety of floor system types, <u>spBeam</u> is equipped to provide cost-effective, accurate, and fast solutions to engineering challenges.

<u>spBeam</u> provides top and bottom bar details including development lengths and material quantities, as well as live load patterning and immediate and long-term deflection results. Using the moment redistribution feature engineers can deliver safe designs with savings in materials and labor. Engaging this feature allows up to 20% reduction of negative moments over supports reducing reinforcement congestions in these areas.

Beam analysis and design requires engineering judgment in most situations to properly simulate the behavior of the targeted beam and take into account important design considerations such as: designing the beam as rectangular or T-shaped sections; using the effective flange width or the center-to-center distance between the beam and the adjacent beams. Regardless which of these options is selected, <u>spBeam</u> provide users with options and flexibility to:

- 1. Design the beam as a rectangular cross-section or a T-shaped section.
- 2. Use the effective or full beam flange width.
- 3. Include the flanges effects in the deflection calculations.
- 4. Invoke moment redistribution to lower negative moments
- 5. Using gross (uncracked) or effective (cracked) moment of inertia

For illustration and comparison purposes, the following figures provide a sample of the results obtained from an <u>spBeam</u> model created for the cantilever beam discussed in this example.



sp Beam - [C:\StructurePoint\Cantilever RC Beam - ACI - Design.slb Is	ometric View]	- 🗆 X
In the input Solve View Options Window Help In the input Solve View Options Window Help In the input Solve View Options		
General Information Span Control Solve Options Labels Project: RC Cantilever Beam Frame: RC Cantilever Beam Engineer: StructurePoint Options Run mode Design code: ACI 318-14 Reinforcement: ASTM A615 No. of Supports: 2 Left cantilever Right cantilever Other Other Distance location as ratio of span Next > Cancel	Span Data Slabs/Planges Longitudinal Beams Span: Wdth: Depth: 24 Modify Copy Span No. Width 1 16 2 16	Depth 24 24 24
Reinforcement Criteria X	Support Data	×
Slabs and Ribs Beams Cover (in) Clear: 2 Bar size Min: #9 v #9 v Max: #9 v #9 v Max: #4 v	Columns Column Capitals Transverse Beams Bound Support: 1 Image: Column Capitals Height (ft) Stiffness share %: 999 Below: 0	ary Conditions c1 (in) c2 (in) 0 0 0 0
Spacing (in) Spacing (in) Spacing (in) Min: 1 1 Max: 18 Min: 6 Reinf. ratio (%) Number of legs Number of legs Min: 0.14 0.14 Min: 2 Max: 5 5 First Stimup from FOS (in) - Clear distance between 1 Dist: 3	Modify Copy Sup. No Suff% HtA c1A c2A 1 999 0 0 0 2 999 0 0 0	HtB c1B c2B 0 0 0 0 0 0 0 0 0
There is more than 12 in of concrete below top bars.		
OK Cancel		
x		OK Cancel
Z		
Ready	Geometry ft	ACI 318-14







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1. Input Echo

1.1. General Information

File Name	C:\Struc\Cantilever RC Beam - ACI - Design.slb
Project	RC Cantilever Beam
Frame	RC Cantilever Beam
Engineer	StructurePoint
Code	ACI 318-14
Reinforcement Database	ASTM A615
Mode	Design
Number of supports =	2 + Right cantilever
Floor System	One-Way/Beam

1.2. Solve Options

ive load pattern ratio = 0%	
Deflections are based on cracked section properties.	
n negative moment regions, Ig and Mcr DO NOT include flange/slab contribution (if available))
ong-term deflections are calculated for load duration of 60 months.	
0% of live load is sustained.	
Compression reinforcement calculations NOT selected.	
Default incremental rebar design selected.	
Noment redistribution NOT selected.	
Effective flange width calculations NOT selected.	
Rigid beam-column joint NOT selected.	
Forsion analysis and design NOT selected.	

1.3. Material Properties

1.3.1. Concrete: Slabs / Beams

Wc	150	lb/ft ³
f'c	4	ksi
Ec	3834.3	ksi
f,	0.47434	ksi

1.3.2. Concrete: Columns

Wc	150	lb/ft ³
f'c	4	ksi
Ec	3834.3	ksi
fr	0.47434	ksi

1.3.3. Reinforcing Steel

Epoxy coated bars	No	
Es	29000	ksi
f _{yt}	60	ksi
f _y	60	ksi





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1.4. Reinforcement Database

Size	Db	Ab	Wb	Size	Db	Ab	Wb
	in	in²	lb/ft		in	in ²	lb/ft
#3	0.38	0.11	0.38	#4	0.50	0.20	0.67
#5	0.63	0.31	1.04	#6	0.75	0.44	1.50
#7	0.88	0.60	2.04	#8	1.00	0.79	2.67
#9	1.13	1.00	3.40	#10	1.27	1.27	4.30
#11	1.41	1.56	5.31	#14	1.69	2.25	7.65
#18	2.26	4.00	13.60				

1.5. Span Data

1.5.1. Slabs

Span	Loc	L1	t	wL	wR	H _{min}
		ft	in	ft	ft	in
1	Int	1.333	0.00	0.667	0.667	0.00
2	Int	8.333	0.00	0.667	0.667	0.00 RC

1.5.2. Ribs and Longitudinal Beams

Notes: *c - Deep beam. Additional design and bar detailing required.

Span		Ribs		Bea	ms	Span		
	b	h	Sp	b	h	H _{min}		
	in	in	in	in	in	in		
1	0.00	0.00	0.00	16.00	24.00	1.00	*c	
2	0.00	0.00	0.00	16.00	24.00	12.50		

1.6. Support Data

1.6.1. Columns

Support	c1a	c2a	Ha	c1b	c2b	Hb	Red %
	in	in	ft	in	in	ft	
1	0.00	0.00	0.000	0.00	0.00	0.000	999
2	0.00	0.00	0.000	0.00	0.00	0.000	999

1.6.2. Boundary Conditions

Support	Sprin	Ig	Far En	d	
	K z kip/in	K ry kip-in/rad	Above	Below	
1	0	0	Fixed	Fixed	
2	0	0	Fixed	Fixed	

1.7. Load Data

1.7.1. Load Cases and Combinations

-	Case	Dead	Live
	Туре	DEAD	LIVE
	U1	1.200	1.600

1.7.2. Point Forces

Case/Patt	Span	Wa	La		
		kip	ft		
Dead	2	12.00	8.333		
Live	2	12.00	8.333		





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1.8. Reinforcement Criteria

1.8.1. Slabs and Ribs

	Units	Тор В	ars	Bottom Bars		
		Min.	Max.	Min.	Max.	
Bar Size		#5	#8	#5	#8	
Bar spacing	in	1.00	18.00	1.00	18.00	
Reinf ratio	%	0.14	5.00	0.14	5.00	
Clear Cover	in	1.50		1.50		

There is NOT more than 12 in of concrete below top bars.

1.8.2. Beams

	Units	Top B	ars	Bottom	Bars	Stirrups	
		Min.	Max.	Min.	Max.	Min.	Max.
Bar Size		#9	#9	#9	#9	#4	#4
Bar spacing	in	1.00	18.00	1.00	18.00	6.00	18.00
Reinf ratio	%	0.14	5.00	0.14	5.00		
Clear Cover	in	2.00		2.00			
Layer dist.	in	1.00		1.00			
No. of legs						2	6
Side cover	in					1.50	
1st Stirrup	in					3.00	

There is NOT more than 12 in of concrete below top bars.

2. Design Results

2.1. Top Reinforcement

Notes: *3 - Design governed by minimum reinforcement. *5 - Number of bars governed by maximum allowable spacing.

Span	Zone	Width	M _{max}	X _{max}	A _{s,min}	A _{s,max}	A _{s,req}	Sp _{Prov}	Bars	
		ft	k-ft	ft	in ²	in ²	in ²	in		
1	Left	1.33	0.00	0.000	0.000	6.195	0.000	0.000		
	Midspan	1.33	0.00	0.467	0.000	6.195	0.000	0.000		
	Right	1.33	0.00	1.333	0.480	6.195	0.000	3.539	4-#9	*3 *5
2	Left	1.33	279.99	0.000	1.143	6.195	3.159	3.539	4-#9	
	Midspan	1.33	181.99	2.917	1.143	6.195	1.988	3.539	4-#9	*5
	Right	1.33	98.00	5.416	1.143	6.195	1.044	3.539	4-#9	*3 *5

2.2. Top Bar Details

	Left				Continuous		Right			
Span	Bars	Length	Bars	Length	Bars	Length	Bars	Length	Bars	Length
		ft		ft		ft		ft		ft
1							2-#9	1.33	2-#9	1.33
2					4-#9	8.33				

2.3. Top Bar Development Lengths

		Lef	t		Continuous		Right			
Span	Bars	DevLen	Bars	DevLen	Bars	DevLen	Bars	DevLen	Bars	DevLen
		in		in		in		in		in
1							2-#9	12.00	2-#9	12.00





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		Lef	t		Continuous		Right			
Span	Bars	DevLen	Bars	DevLen	Bars	DevLen	Bars	DevLen	Bars	DevLen
		in		in		in		in		in
2	auro di				4-#9	25.43				

2.4. Bottom Reinforcement

Ī	Span	Width	M _{max}	X _{max}	A _{s,min}	A _{s,max}	A _{s,req}	Sp _{Prov}	Bars	
		ft	k-ft	ft	in ²	in²	in²	in		
	1	1.33	0.00	1.333	0.000	6.195	0.000	0.000		
	2	1.33	0.00	4.167	0.000	6.195	0.000	0.000		

2.5. Bottom Bar Details

	L	ong Ba	ars	Short Bars				
Span	Bars	Start	Length	Bars	Start	Length		
		ft	ft		ft	ft		
1								
2								

2.6. Bottom Bar Development Lengths

	Lon	ng Bars	Short Bars		
Span	Bars	DevLen in	Bars	DevLen in	
1					
2					

2.7. Flexural Capacity

				Тор					Botto	m	
Span	x	$A_{s,top}$	ФМ _n -	M _u -	Comb Pat	Status	A _{s,bot}	ФМ _n +	M _u +	Comb Pat	Status
	ft	in ²	k-ft	k-ft			in ²	k-ft	k-ft		
1	0.000	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	0.467	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	0.666	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	0.866	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	1.000	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	1.333	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
2	0.000	4.00	-346.14	-279.99	U1 All	ок	0.00	0.00	0.00	U1 All	OK
	2.917	4.00	-346.14	-181.99	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	4.167	4.00	-346.14	-139.99	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	5.416	4.00	-346.14	-98.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK
	8.333	4.00	-346.14	0.00	U1 All	OK	0.00	0.00	0.00	U1 All	OK

2.8. Longitudinal Beam Transverse Reinforcement Demand and Capacity

2.8.1. Section Properties

ΦV。	(A _v /s) _{min}	d	Span
kip	in²/in	in	
32.54	0.0133	21.44	1
32.54	0.0133	21.44	2





2.8.2. Beam Transverse Reinforcement Demand Notes: *8 - Minimum transverse (stirrup) reinforcement governs.

				Demand				
Span	Start	End	Xu	Vu	Comb/Patt	A _v /s	A _v /s	
	ft	ft	ft	kip		in²/in	in²/in	
1	0.250	1.083	0.666	0.00	U1/All	0.0000	0.0000	
2	0.250	3.969	1.786	33.60	U1/All	0.0011	0.0133	*8
	3.969	6.151	3.969	33.60	U1/All	0.0011	0.0133	*8
	6.151	8.333	6.151	33.60	U1/All	0.0011	0.0133	*8

2.8.3. Beam Transverse Reinforcement Details

Span Size Stirrups (2 legs each unless otherwise noted)

1 #4 --- None --

2 #4 10@10.4

2.8.4. Beam Transverse Reinforcement Capacity

Notes: *8 - Minimum transverse (stirrup) reinforcement governs.

				Re	quired			F	Provided		
Span	Start	End	Xu	Vu	Comb/Patt	A _v /s	Av	Sp	A _v /s	ΦVn	
	ft	ft	ft	kip		in²/in	in ²	in	in²/in	kip	
1	0.000	1.333	0.666	0.00	U1/All	0.0000				16.27	
2	0.000	0.250	1.786	33.60	U1/All						
	0.250	8.083	1.786	33.60	U1/All	0.0011	0.40	10.4	0.0383	69.48 *8	ŝ
	8.083	8.333	8.083	33.60	U1/All						

2.9. Slab Shear Capacity

Span	b	d	V _{ratio}	ΦVc	Vu	Xu
	in	in		kip	kip	ft
1 1	Not checke	ed				
2 1	Not checke	ed				

2.10. Material TakeOff

2.10.1. Reinforcement in the Direction of Analysis

Top Bars	131.5 lb	<=>	13.60 lb/ft	<=>	10.200 lb/ft2
Bottom Bars	0.0 lb	<=>	0.00 lb/ft	<=>	0.000 lb/ft2
Stirrups	37.9 lb	<=>	3.92 lb/ft	<=>	2.937 lb/ft2
Total Steel	169.3 lb	<=>	17.52 lb/ft	<=>	13.137 lb/ft2
Concrete	25.8 ft ³	<=>	2.67 ft3/ft	<=>	2.000 ft3/ft2

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3. Deflection Results: Summary 3.1. Section Properties

3.1.1. Frame Section Properties

Notes:

M-ve values are for positive moments (tension at bottom face). M-ve values are for negative moments (tension at top face).

	M	tve		I	И. _{ve}	
Span Zone	lg	l _{cr}	Mcr	l _g	I _{cr}	M _{cr}
	in ⁴	in ⁴	k-ft	in ⁴	in ⁴	k-ft
1 Left	18432	0	60.72	18432	0	-60.72
Midspan	18432	0	60.72	18432	8120	-60.72
Right	18432	0	60.72	18432	8120	-60.72
2 Left	18432	0	60.72	18432	8120	-60.72
Midspan	18432	0	60.72	18432	8120	-60.72
Right	18432	0	60.72	18432	8120	-60.72

3.1.2. Frame Effective Section Properties

		Load Level							
		Dead		Sustaine	Sustained		ve		
Span Zone	Weight	M _{max}	I.	M _{max}	I.	M _{max}	l,		
		k-ft	in ⁴	k-ft	in ⁴	k-ft	in ⁴		
1 Left	0.150	0.00	18432	0.00	18432	0.00	18432		
Middle	0.700	0.00	18432	0.00	18432	0.00	18432		
Right	0.150	0.00	18432	0.00	18432	0.00	18432		
Span Avg			18432		18432		18432		
2 Left	1.000	-100.00	10428	-100.00	10428	-199.99	8409		
Span Avg			10428		10428		8409		

3.2. Instantaneous Deflections

3.2.1. Extreme Instantaneous Frame Deflections and Corresponding Locations

						Live		Tota	al
Span	Direction	Value	Units	Dead	Sustained	Unsustained	Total	Sustained	Dead+Live
1	Down	Def	in						
		Loc	ft						
	Up	Def	in						
		Loc	ft						
2	Down	Def	in	0.100		0.148	0.148	0.100	0.248
		Loc	ft	8.333		8.333	8.333	8.333	8.333
	Up	Def	in						1.000
		Loc	ft						

3.3. Long-term Deflections

3.3.1. Long-term Deflection Factors

Notes: Deflection multiplier, Lambda, depends on moment sign at sustained load level and Rho' in given zone. Rho' is assumed zero because Compression Reinforcement option is NOT selected in Solve Options.

Time dependant factor for sustained loads = 2.000

			M+ve					M.,,	1	
Span Zone	A _{s,top}	b	d in	Rho' %	Lambda	A _{s,bot}	b	d in	Rho' %	Lambda
1 Midspan				0.000	2.000				0.000	2.000
2 Left				0.000	2.000				0.000	2.000





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3.3.2. Extreme Long-term Frame Deflections and Corresponding Locations Notes:

Notes:
Incremental deflections due to creep and shrinkage (cs) based on sustained load level values.
Incremental deflections after partitions are installed can be estimated by deflections due to:
creep and shrinkage plus unsustained live load (cs+lu), if live load applied before partitions,
creep and shrinkage plus live load (cs+l), if live load applied after partitions.
Total deflections consist of dead, live, and creep and shrinkage deflections.

Tota	cs+l	cs+lu	CS	Units	Value	Direction	Span
				in	Def	Down	1
				ft	Loc		
				in	Def	Up	
				ft	Loc		
0.44	0.348	0.348	0.200	in	Def	Down	2
8.33	8.333	8.333	8.333	ft	Loc		
				in	Def	Up	
				ft	Loc		





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4. Diagrams 4.1. Loads

	12 kip
	.u
	12 kip
CASE: Dea	d
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Project: RC Cantilever Beam	
Frame: RC Cantilever Beam	
Engineer: StructurePoint	
Code: ACI 318-14	
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4.2. Internal Forces







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4.3. Moment Capacity







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4.4. Shear Capacity









4.5. Deflection







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4.6. Reinforcement







8. Analysis and Design Results Comparison and Conclusions

The following tables show the comparison between hand results and <u>spBeam</u> model results.

Table 2 - Comparison of Moments and Flexural Reinforcement (At Fixed End)							
Location	M _u , kip-ft	$A_{s,required}$, in. ²	$A_{s,min}$, in. ²	Reinforcement	S _{provided} , in.	$A_{s,provided}$, in. ²	
Hand	280.00	3.160	1.143	4 – #9	3.539	4.000	
<u>spBeam</u>	279.99	3.159	1.143	4 – #9	3.539	4.000	

Table 3 - Comparison of Shear and lateral Reinforcement									
V _{u@}	d, kip	(A _v in	/s) _{req} *, .²/ in.	$(A_v/s)_{min}^*$, Reinforce $in.^{2/}$ in.		cement ϕV_n , kip		n, kip	
Hand	<u>spBeam</u>	Hand	<u>spBeam</u>	Hand	<u>spBeam</u>	Hand	<u>spBeam</u>	Hand	<u>spBeam</u>
33.6	33.6	0.0011	0.0011	0.0133	0.0133	10 - #4 @ 10.444 in.	10 - #4 @ 10.444 in.	69.48	69.48
* Minimum transverse reinforcement governs									

Table 4 - Comparison of Section Properties I_{cr}, in.⁴ Ie, in.⁴ Location Hand <u>spBeam</u> Hand <u>spBeam</u> DL DL+LL_{sus} Total DL DL+LL_{sus} Total Midspan 8120 8120 10428 10428 8409 10428 10428 8409

Table 5 - Comparison of Maximum Instantaneous Deflection (At Free End), in.						
Deflection Type	Hand	<u>spBeam</u>				
$\Delta_{ m DL}$	0.100	0.100				
Δ_{LL}	0.148	0.148				
$\Delta_{ m total}$	0.248	0.248				

Table 6 - Comparison of Maximum Long-Term Deflection (At Free End), in.						
Deflection Type	Hand	<u>spBeam</u>				
$\Delta_{ m cs}$	0.200	0.200				
$\Delta_{ m cs}+\Delta_{ m LL}$	0.348	0.348				
$(\Delta_{\text{total}})_{\text{lt}}$	0.448	0.448				

The results of all the hand calculations used illustrated above are in agreement with the automated exact results obtained from the <u>spBeam</u> program.