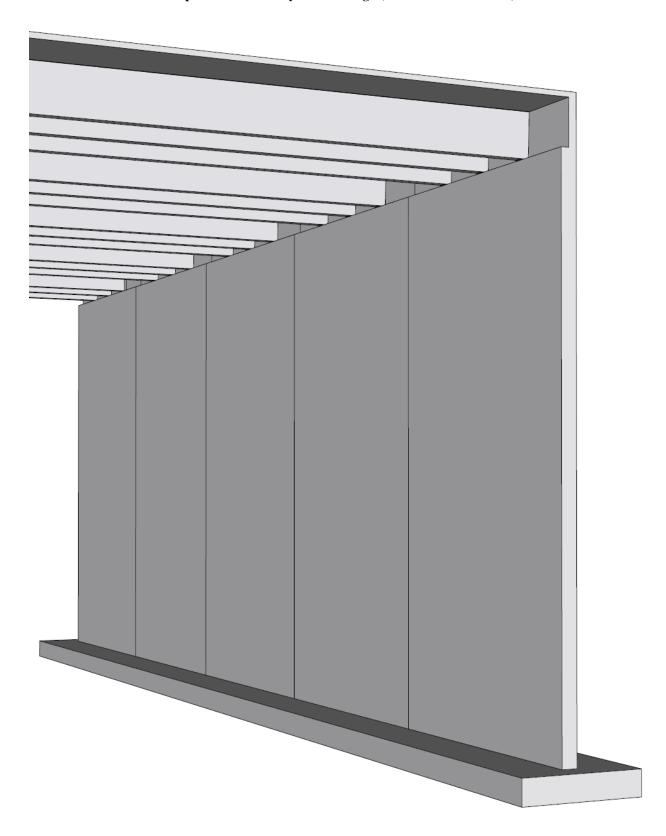




Reinforced Concrete Tilt-Up Wall Panel Analysis and Design (ACI 318-14 – ACI 551)







Reinforced Concrete Tilt-Up Wall Panel Analysis and Design (ACI 318-14 - ACI 551)

Tilt-up is form of construction with increasing popularity owing to its flexibility and economics. Tilt-up concrete is essentially a precast concrete that is site cast instead of traditional factory cast concrete members. A structural reinforced concrete tilt-up wall panel in a single-story warehouse (big-box) building provides gravity and lateral load resistance for the following applied loads from three roof joists bearing in wall pockets in addition to the wind:

Roof dead load = 2.4 kip per joist Roof live load = 2.5 kip per joist

Wind load = 27.2 psf (Out-of-Plane)

The assumed tilt-up wall panel section and reinforcement are investigated after analysis to verify suitability for the applied loads then compared with numerical analysis results obtained from <u>spWall</u> engineering software program from <u>StructurePoint</u>.

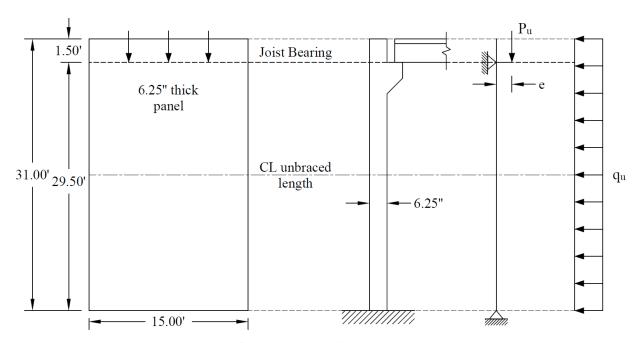


Figure 1 – Reinforced Concrete Tilt-Up Wall Panel Geometry

Version: Jan-30-2023





Contents

| 1. | Minimum Vertical Reinforcement | 2 |
|----|--|---|
| | Alternative Method for Out-of-Plane Slender Wall Analysis ACI 318 Provisions | |
| | Tilt-Up Wall Structural Analysis | |
| ٥. | | |
| | 3.1. Applied loads | |
| | 3.2. Maximum wall forces | |
| | 3.3. Tension-controlled verification | 4 |
| 4. | Tilt-Up Wall Cracking Moment Capacity (M _{cr}) | 5 |
| 5. | $Tilt-Up \ Wall \ Flexural \ Moment \ Capacity \ (\phi M_n)$ | 5 |
| 6. | Tilt-Up Wall Vertical Stress Check | 5 |
| 7. | Tilt-Up Wall Shear Stress Check | 5 |
| 8. | Tilt-Up Wall Mid-Height Deflection (Δ _s) | 5 |
| | Tilt-Up Wall Panel Analysis and Design – spWall Software | |
| | Design Results Comparison and Conclusions | |





Code

Building Code Requirements for Structural Concrete (ACI 318-14) and Commentary (ACI 318R-14)

Reference

- Design Guide for Tilt-Up Concrete Panels, ACI 551.2R-15, 2015, Example B.1
- spWall Engineering Software Program Manual v10.00, STRUCTUREPOINT, 2022

Design Data

```
f_c' = 4,000 psi normal weight concrete (w<sub>c</sub> = 150 pcf)

f_y = 60,000 psi

Wall length = l_c = 31 ft – 1.5 ft = 29.5 ft

Assumed wall thickness = 6.25 in.

Assumed eccentricity = e_{cc} = 3 in.

Assumed vertical reinforcement: 16 #6 (single layer)
```





1. Minimum Vertical Reinforcement

$$\rho_{l} = \frac{A_{v,vertical}}{b \times h} = \frac{7.04}{(15 \times 12) \times 6.25} = 0.0063$$
ACI 318-14 (2.2)

$$\rho_{l,\text{min}} = 0.0015$$
ACI 318-14 (Table 11.6.1)

$$\rho_l = 0.0063 \ge \rho_{l,\text{min}} = 0.0015 \text{ (o.k)}$$

$$s_{l,\text{max}} = \text{smallest of} \begin{cases} 3 \times h \\ 18 \text{ in.} \end{cases} = \text{smallest of} \begin{cases} 3 \times 6.25 \\ 18 \text{ in.} \end{cases} = \text{smallest of} \begin{cases} 18.75 \text{ in.} \\ 18 \text{ in.} \end{cases} = 18 \text{ in.}$$

$$\frac{ACI 318-14 (11.7.2.1)}{18 \text{ in.}} = 18 \text{ in.}$$

$$s_{l,provided} = \frac{15 \times 12}{16} = 11.25 \text{ in.} \le s_{l,max} = 18 \text{ in. } (\textbf{o.k.})$$

2. Alternative Method for Out-of-Plane Slender Wall Analysis ACI 318 Provisions

The design guide for tilt-up concrete panels ACI 551 states that tilt-up concrete walls can be analyzed using the provisions of Chapter 14 of the ACI 318-11, the same provisions are presented in Chapter 11 of the ACI 318-14. Most walls, and especially slender walls, are widely evaluated using the "Alternative Method for Out-of-Plane Slender Wall Analysis" in Section 11.8 of the ACI 318-14. The method is applicable when the conditions summarized below are met:

| • | The cross section shall be constant over the height of the wall | <u>ACI 318-14 (11.8.1.1(a))</u> |
|---|---|---------------------------------|
| • | The wall can be designed as simply supported | <u>ACI 318-14 (11.8.2.1)</u> |
| • | Maximum moments and deflections occurring at midspan | <u>ACI 318-14 (11.8.2.1)</u> |
| • | The wall must be axially loaded | ACI 318-14 (11.8.2.1) |
| • | The wall must be subjected to an out-of-plane uniform lateral load | ACI 318-14 (11.8.2.1) |
| • | The wall shall be tension-controlled | ACI 318-14 (11.8.1.1(b)) |
| • | The reinforcement shall provide design strength greater than cracking strength | <u>ACI 318-14 (11.8.1.1(c))</u> |
| • | P_u at the midheight section does not exceed $0.06f_c$ A_g | <u>ACI 318-14 (11.8.1.1(d))</u> |
| • | Out-of-plane deflection due to service loads including Pa effects does not exce | $eed l_c/150$ |

ACI 318-14 (11.8.1.1(e))





3. Tilt-Up Wall Structural Analysis

Using 11.8 provisions, calculate factored loads as follows for each of the considered load combinations:

3.1. Applied loads

Wall self-weight =
$$\frac{6.25}{12} \times 15 \times \left(\frac{29.5}{2} + 1.5\right) \times 150 \times \frac{1 \text{ kip}}{1000 \text{ lb}} = 19.04 \text{ kip}$$

$$P_{DL} = 3 \times 2.4 = 7.20 \text{ kip}$$

$$P_{LL} = 3 \times 2.5 = 7.50 \text{ kip}$$

$$w = 27.2 \text{ lb/ft}^2$$

3.2. Maximum wall forces

Calculate maximum factored wall forces in accordance with 11.8.3.1 including moment magnification due to second order (P- Δ) effects. Load combination U = 1.2 D + 1.6 L_r + 0.5 W is considered in this example:

$$P_{ua} = 1.2 \times 7.20 + 1.6 \times 7.50 = 20.64 \text{ kip}$$

$$P_{um} = 20.64 + 1.2 \times 19.04 = 43.49 \text{ kip}$$

$$w_u = 0.5 \times 27.2 \times 15 \text{ ft} \times \frac{1 \text{ kip}}{1000 \text{ lb}} = 0.204 \text{ kip/ft}$$

$$M_{u} = \frac{M_{ua}}{1 - \frac{5 \times P_{u} \times l_{c}^{2}}{0.75 \times 48 \times E_{c} \times I_{cr}}}$$

$$\underline{ACI 318-14 (Eq. 11.8.3.1(d))}$$

$$M_{ua} = \frac{w_u \times l_c^2}{8} + \frac{P_{ua} \times e}{2} = \frac{0.204 \times (29.5)^2}{8} + \frac{20.64 \times 3}{2 \times 12} = 24.77 \text{ ft-kip}$$

Where M_{ua} is the maximum factored moment at midheight of wall due to lateral and eccentric vertical loads, not including $P\Delta$ effects. ACI 318-14 (11.8.3.1)

$$E_c = 57,000 \times \sqrt{f_c} = 57,000 \times \sqrt{4,000} = 3,605,000 \text{ psi}$$

$$\underline{ACI 318-14 (19.2.2.1(b))}$$

$$I_{cr} = n \times A_{se} \times (d - c)^2 + \frac{l_w \times c^3}{3}$$
ACI 318-14 (11.8.3.1(c))

$$n = \frac{E_s}{E_s} = \frac{29,000}{3,605} = 8.0 > 6.0 \text{ (o.k.)}$$
ACI 318-14 (11.8.3.1)

Calculate the effective area of longitudinal reinforcement in a slender wall for obtaining an approximate cracked moment of inertia.





$$A_{se} = A_s + \frac{P_{um} \times h}{2 \times f_v \times d} = 7.04 + \frac{43.49 \times 6.25}{2 \times 60 \times (6.25/2)} = 7.76 \text{ in.}^2$$

$$\underline{ACI 318-14 (R11.8.3.1)}$$

The following calculation are performed with the effective area of steel in lieu of the actual area of steel.

$$a = \frac{A_{se} \times f_y}{0.85 \times f_c \times b} = \frac{7.76 \times 60}{0.85 \times 4 \times (15 \times 12)} = 0.761 \text{ in.}$$

$$c = \frac{a}{\beta} = \frac{0.761}{0.85} = 0.896$$
 in.

$$\frac{c}{d} = \frac{0.896}{3.13} = 0.287 < 0.375$$
 : tension-controlled

ACI 318-14 (R21.2.2)

$$\phi = 0.9$$

ACI 318-14 (Table 21.2.2)

$$I_{cr} = 8.0 \times 7.76 \times (3.13 - 0.896)^2 + \frac{(15 \times 12) \times 0.896^3}{3} = 353.56 \text{ in.}^4$$

ACI 318-14 (11.8.3.1(c))

$$M_{u} = \frac{M_{ua}}{1 - \frac{P_{um}}{0.75 \times K_{b}}}$$

ACI 318-14 (Eq. 11.8.3.1(d))

$$K_b = \frac{48 \times E_c \times I_{cr}}{5 \times l_c^2} = \frac{48 \times 3605 \times 353.56}{5 \times (29.5 \times 12)^2} = 97.64 \text{ kip}$$

$$M_u = \frac{24.77}{1 - \frac{43.49}{0.75 \times 97.64}} = 61.00 \text{ ft-kip}$$

3.3. Tension-controlled verification

ACI 318-14 (11.8.1.1(b))

$$P_n = \frac{P_{um}}{\phi} = \frac{43.49}{0.9} = 48.32 \text{ kips}$$

$$a = \frac{A_{se,w} \times f_y}{0.85 \times f_c \times l_w} = \frac{\frac{P_n \times h}{2 \times d} + A_s \times f_y}{0.85 \times f_c \times l_w} = \frac{\frac{48.32 \times 6.25}{2 \times 3.13} + 7.04 \times 60}{0.85 \times 4 \times 15 \times 12} = 0.769 \text{ in.}$$

$$c = \frac{a}{\beta_1} = \frac{0.769}{0.85} = 0.905$$
 in.

$$\varepsilon_{t} = \left(\frac{0.003}{c}\right) \times d_{t} - 0.003 = \left(\frac{0.003}{0.905}\right) \times 3.13 - 0.003 = 0.0075 > 0.0050$$

Therefore, section is tension controlled

ACI 318-14 (Table 21.2.2)



ACI 318-14 (11.5.1.1(b))



4. Tilt-Up Wall Cracking Moment Capacity (Mcr)

Determine f_r = Modulus of rapture of concrete and I_g = Moment of inertia of the gross uncracked concrete section to calculate M_{cr}

$$f_r = 7.5\lambda\sqrt{f_c'} = 7.5 \times 1.0 \times \sqrt{4,000} = 474.3 \text{ psi}$$

$$I_g = \frac{l_w h^3}{12} = \frac{(15 \times 12) \times 6.25^3}{12} = 3662 \text{ in.}^4$$

$$y_t = \frac{h}{2} = \frac{6.25}{2} = 3.13 \text{ in.}$$

$$M_{cr} = \frac{f_r I_g}{v_r} = \frac{474.3 \times 3662}{3.13} \times \frac{1}{1000} \times \frac{1}{12} = 46.32 \text{ ft-kip}$$

$$ACI 318-14 (24.2.3.5(b))$$

5. Tilt-Up Wall Flexural Moment Capacity (\$\phi M_n\$)

For load combination #1:

$$M_n = A_{se} \times f_y \times \left(d - \frac{a}{2}\right) = 7.76 \times 60 \times \left(3.13 - \frac{0.761}{2}\right) = 1278.58 \text{ in.-kip} = 106.55 \text{ ft-kip}$$

It was shown previously that the section is tension controlled $\rightarrow \phi = 0.9$

 $\phi M_n = \phi \times M_n = 0.9 \times 106.55 = 95.89 \text{ ft-kip} > M_u = 61.00 \text{ ft-kip} \text{ (o.k.)}$

$$\phi M_n = 95.89 \text{ ft-kip} > M_{cr} = 46.32 \text{ ft-kip } (\mathbf{o.k.})$$

$$\Delta_u = \frac{M_u}{0.75 \times K_h} = \frac{61.00 \times 12}{0.75 \times 97.64} = 9.995 \text{ in.}$$

$$\underline{ACI 318-14 (11.8.3.1(b))}$$

6. Tilt-Up Wall Vertical Stress Check

$$\frac{P_{um}}{A_c} = \frac{43.49 \times 1000}{6.25 \times (15 \times 12)} = 38.66 \text{ psi} < 0.06 \times f_c' = 0.06 \times 4,000 = 240 \text{ psi} \text{ (o.k.)}$$

$$\underline{ACI 318-14 (11.8.1.1(d))}$$

7. Tilt-Up Wall Shear Stress Check

In-plane shear is not evaluated since in-plane shear forces are not applied in this example. Out-of-plane shear due to lateral load should be checked against the shear capacity of the wall. By inspection of the maximum shear forces, it can be determined that the maximum shear force is under 10 kips. The wall has a shear capacity approximately 140 kips and no detailed calculations are required by engineering judgement. (See Figure 10 for detailed shear force diagram)

8. Tilt-Up Wall Mid-Height Deflection (Δs)

The maximum out-of-plane deflection (Δ_s) due to service lateral and eccentric vertical loads, including P Δ effects, shall not exceed $l_c/150$. Where Δ_s is calculated as follows:

ACI 318-14 (11.8.1.1(e))





$$\Delta_{s} = \begin{cases} \frac{2}{3} \Delta_{cr} + \frac{M_{a} - \frac{2}{3} M_{cr}}{M_{n} - \frac{2}{3} M_{cr}} \times \left(\Delta_{n} - \frac{2}{3} \Delta_{cr}\right) & \text{When} & M_{a} > \frac{2}{3} M_{cr} \\ \left(\frac{M_{a}}{M_{cr}}\right) \Delta_{cr} & \text{When} & M_{a} < \frac{2}{3} M_{cr} \end{cases}$$

$$\frac{ACI 318-14 (Table 11.8.4.1)}{M_{cr}}$$

Where M_a is the maximum moment at mid-height of wall due to service lateral and eccentric vertical loads including $P\Delta$ effects.

$$M_a = M_{sa} + P_s \Delta_s$$

$$M_{sa} = \frac{w_s \times l_c^2}{8} + \frac{P_a \times e}{2} = \frac{\left(0.7 \times \frac{27.2}{1.6} \times 15\right) \times \left(29.5\right)^2}{8 \times 1000} + \frac{\left(7.20\right) \times 3/12}{2} = 20.32 \text{ ft-kip}$$

 $P_s = P_{DL}$ + wall self-weight = 7.20 + 19.04 = 26.24 kip

$$M_{cr} = \frac{f_r I_g}{y_t} = 46.32 \text{ ft-kip (as calculated previously)}$$

$$\underline{ACI 318-14 (24.2.3.5(b))}$$

$$\Delta_{cr} = \frac{5}{48} \times \frac{M_{cr} \times l_c^2}{E_c \times I_a} = \frac{5}{48} \times \frac{46.32 \times 12 \times (29.5 \times 12)^2}{3,605 \times 3662} = 0.550 \text{ in.}$$

$$\underline{ACI 318-14 (11.8.4.3(a))}$$

 Δ_s will be calculated by trial-and-error method since Δ_s is a function of M_a and M_a is a function of Δ_s .

Assume
$$M_{sa} < \frac{2}{3} M_{cr}$$

Assume
$$\Delta_s = \left(\frac{M_{sa}}{M_{cr}}\right) \Delta_{cr} = \left(\frac{20.32}{46.32}\right) \times 0.550 = 0.241 \text{ in.}$$

$$M_a = M_{sa} + P_s \Delta_s = 20.32 \times 12 + 26.24 \times 0.241 = 250.14 \text{ in.-kip} = 20.84 \text{ ft-kip}$$

$$\Delta_s = \left(\frac{M_a}{M_{cr}}\right) \Delta_{cr} = \frac{20.84}{46.32} \times 0.550 = 0.247 \text{ in.}$$
ACI 318-14 (Table 11.8.4.1)

No further iterations are required.

$$M_a = 20.84 \text{ ft-kip} < \frac{2}{3} M_{cr} = \frac{2}{3} \times 46.32 = 30.88 \text{ ft-kip}$$
 (o.k.)

$$\Delta_s = 0.247 \text{ in.} < \frac{l_c}{150} = \frac{29.5 \times 12}{150} = 2.36 \text{ in.}$$
 (o.k.)

The wall is adequate with 16 #6 vertical reinforcement and 6.25 in. thickness.





9. Tilt-Up Wall Panel Analysis and Design - spWall Software

<u>spWall</u> is a program for the analysis and design of reinforced concrete shear walls, tilt-up walls, precast walls and Insulate Concrete Form (ICF) walls. It uses a graphical interface that enables the user to easily generate complex wall models. Graphical user interface is provided for:

- Wall geometry (including any number of openings and stiffeners)
- Material properties including cracking coefficients
- Wall loads (point, line, and area)
- Support conditions (including translational and rotational spring supports)

spWall uses the Finite Element Method for the structural modeling, analysis, and design of slender and non-slender reinforced concrete walls subject to static loading conditions. The wall is idealized as a mesh of rectangular plate elements and straight-line stiffener elements. Walls of irregular geometry are idealized to conform to geometry with rectangular boundaries. Plate and stiffener properties can vary from one element to another but are assumed by the program to be uniform within each element.

Six degrees of freedom exist at each node: three translations and three rotations relating to the three Cartesian axes. An external load can exist in the direction of each of the degrees of freedom. Sufficient number of nodal degrees of freedom should be restrained in order to achieve stability of the model. The program assembles the global stiffness matrix and load vectors for the finite element model. Then, it solves the equilibrium equations to obtain deflections and rotations at each node. Finally, the program calculates the internal forces and internal moments in each element. At the user's option, the program can perform second order analysis. In this case, the program takes into account the effect of in-plane forces on the out-of-plane deflection with any number of openings and stiffeners.

In <u>spWall</u>, the required flexural reinforcement is computed based on the selected design standard (ACI 318-14 is used in this example), and the user can specify one or two layers of wall reinforcement. In stiffeners and boundary elements, <u>spWall</u> calculates the required shear and torsion steel reinforcement. Wall concrete strength (in-plane and out-of-plane) is calculated for the applied loads and compared with the code permissible shear capacity.

For illustration and comparison purposes, the following figures provide a sample of the input modules and results obtained from an <u>spWall</u> model created for the reinforced concrete wall in this example.





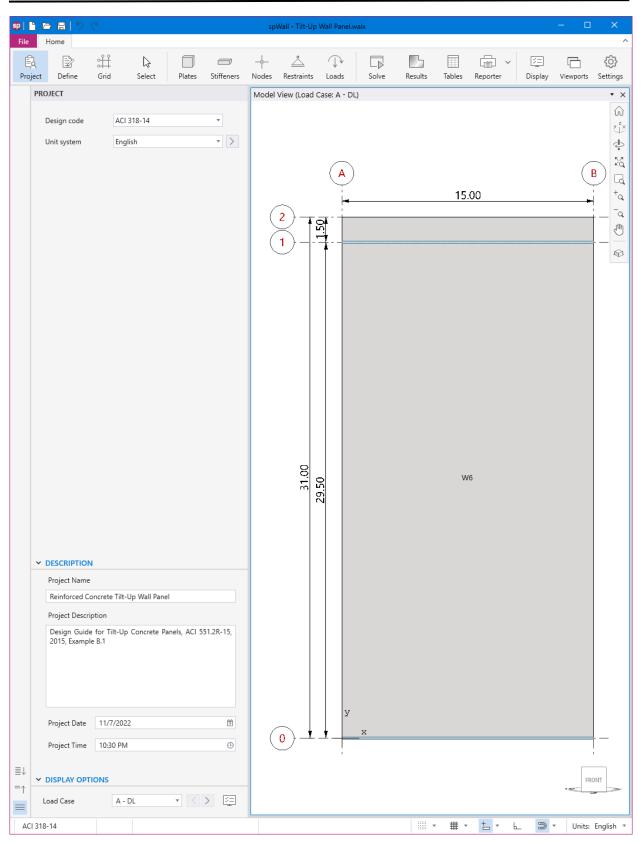


Figure 2 – spWall Interface





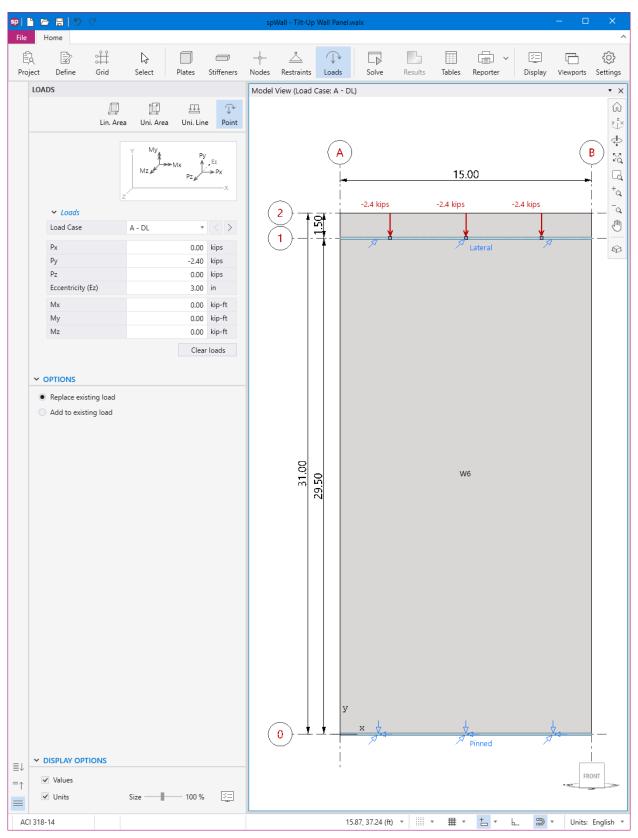


Figure 3 – Assigning Roof Dead Loads for Tilt-Up Wall (spWall)





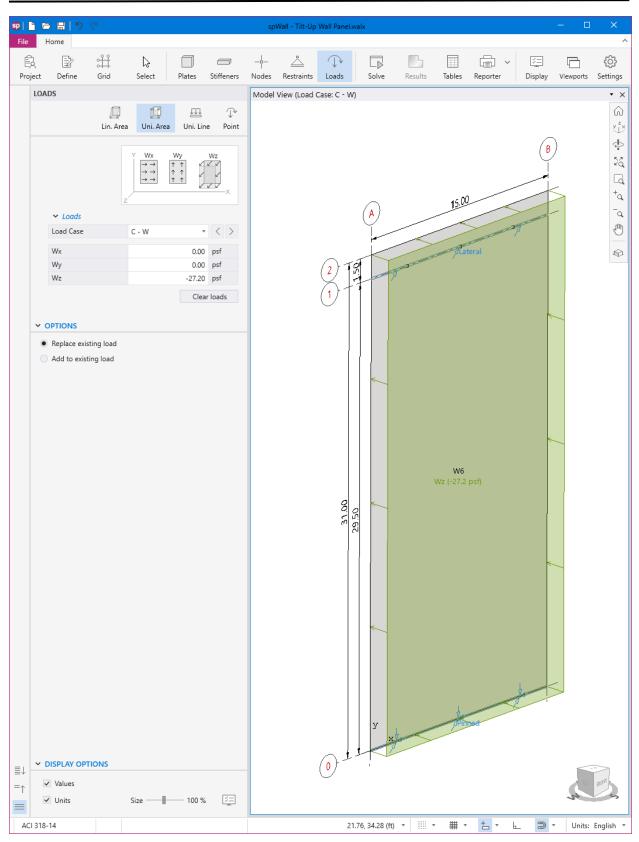


Figure 4 – Assigning Wind Loads for Tilt-Up Wall (spWall)





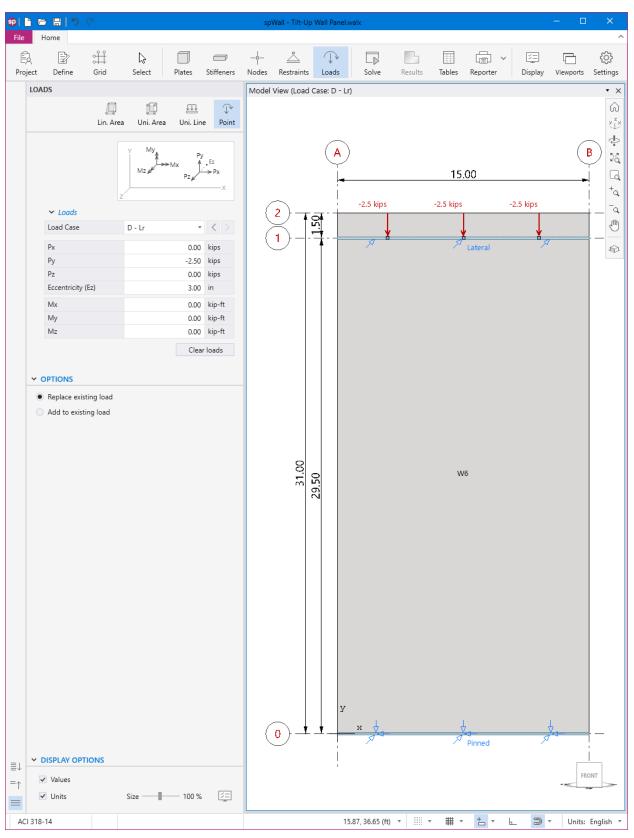


Figure 5 – Assigning Roof Live Loads for Tilt-Up Wall (spWall)





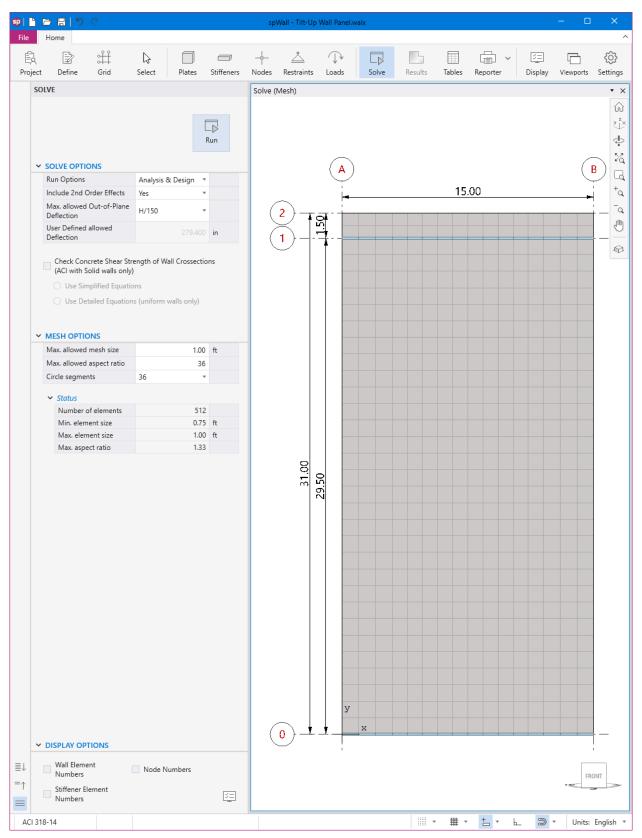


Figure 6 – Solve and Mesh Options (spWall)





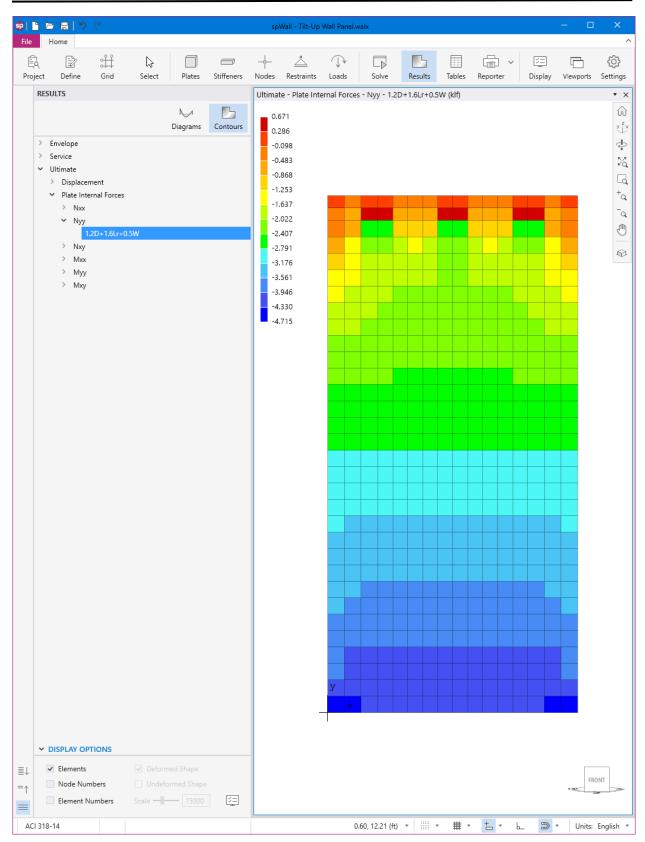


Figure 7 – Factored Axial Forces Contour Normal to Tilt-Up Wall Panel Cross-Section (spWall)





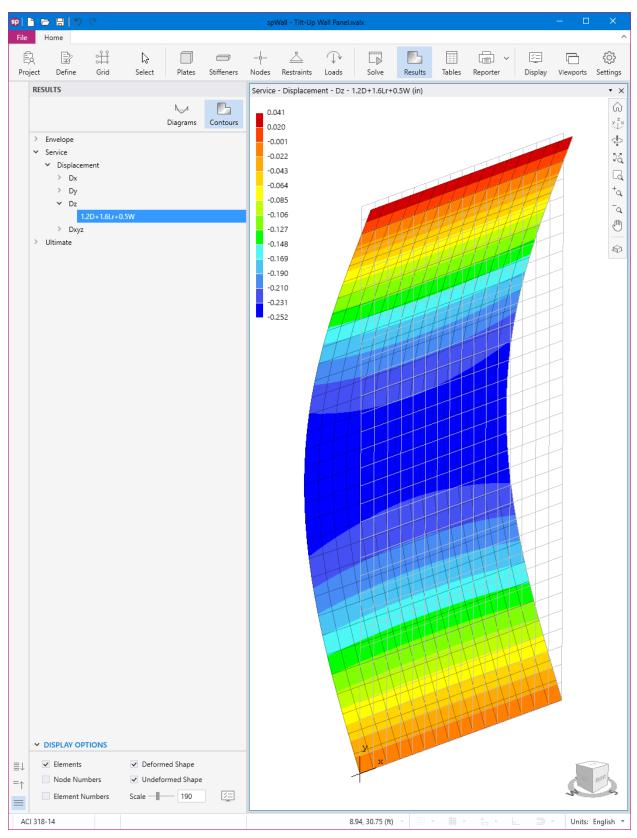


Figure 8 – Tilt-Up Wall Panel Lateral Displacement Contour (Out-of-Plane) (spWall)





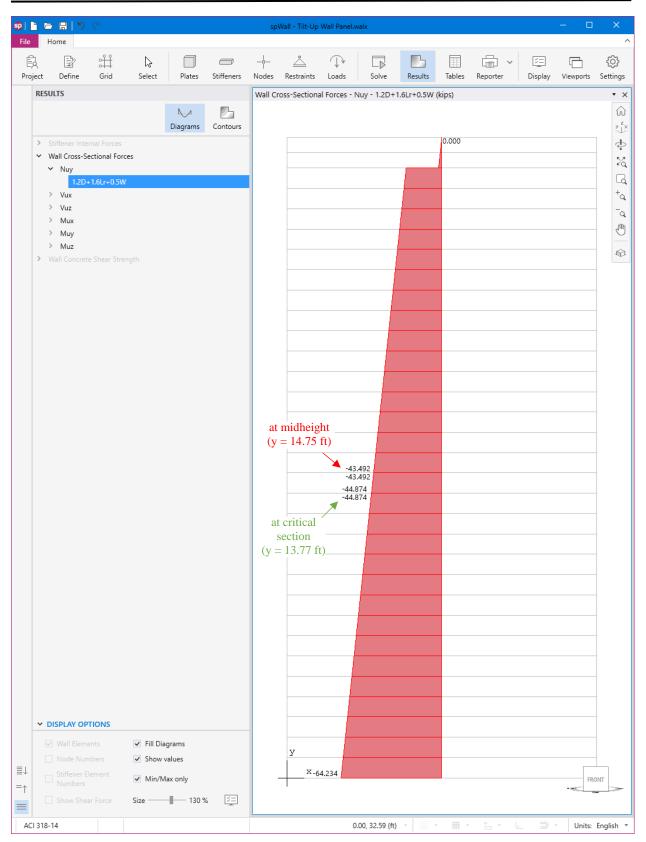


Figure 9 - Tilt-Up Wall Panel Axial Load Diagram (spWall)





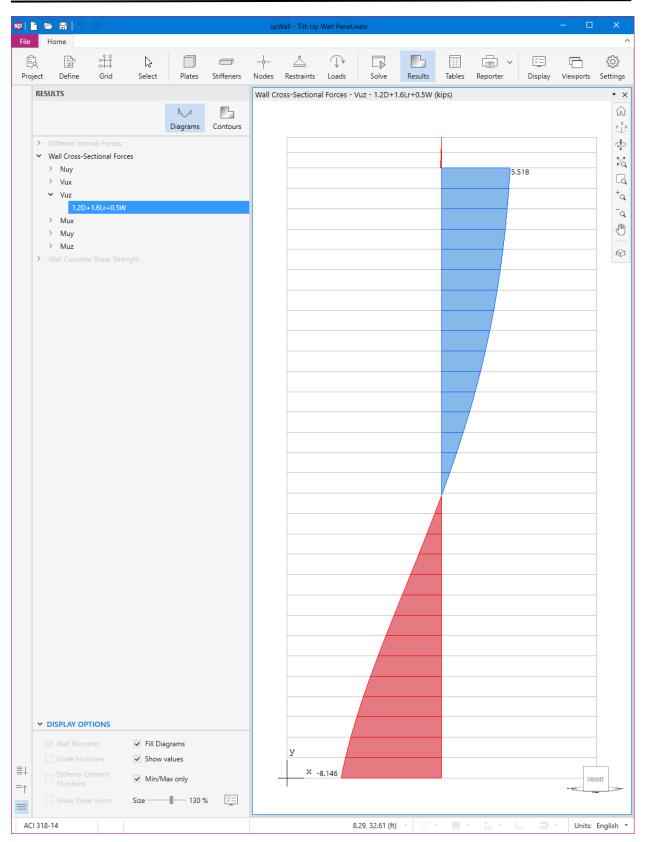


Figure 10 – Out-of-plane Shear Diagram (spWall)





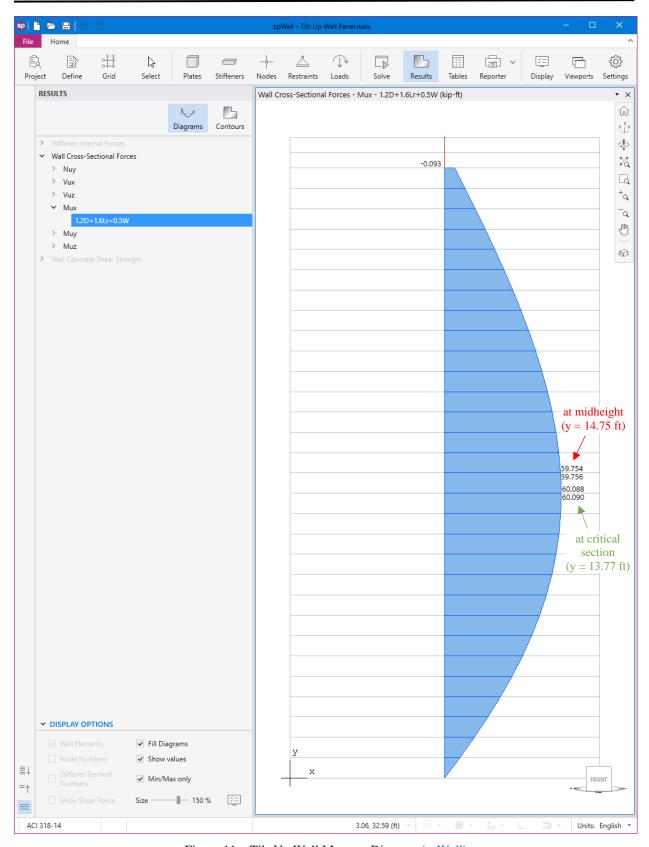


Figure 11 – Tilt-Up Wall Moment Diagram (spWall)







spWall v10.00 (TM)
A Computer Program for Analysis and Design of Reinforced Concrete, Precast, and Tilt-up Walls
Copyright © 1988-2022, STRUCTUREPOINT, LLC.
All rights reserved



Structure Point

Licensee stated above acknowledges that STRUCTUREPOINT (SP) is not and cannot be responsible for either the accuracy or adequacy of the material supplied as input for processing by the spWall computer program. Furthermore, STRUCTUREPOINT neither makes any warranty expressed nor implied with respect to the correctness of the output prepared by the spWall program. Although STRUCTUREPOINT has endeavored to produce spWall error free the program is not and cannot be certified infallible. The final and only responsibility for analysis, design and engineering documents is the licensee's. Accordingly, STRUCTUREPOINT disclaims all responsibility in contract, negligence or other tort for any analysis, design or engineering documents prepared in connection with the use of the spWall program. Licensed to: StructurePoint, LLC. License ID: 00000-0000004-20FC1-20FC1





| STRUCTUREPOINT - spWall v10.00 (TM) |
|--|
| Licensed to: StructurePoint, LLC. License ID: 00000-0000000-4-20FC1-20FC |
| E:\StructurePoint\spWall\Tilt-Up Wall Panel.walx |

Page | 2 1/27/2023 4:11 PM

Contents

| 1. | Project | 3 |
|----|------------------------------------|-----|
| | 1.1. General Information | 3 |
| | 1.2. Solver Options | |
| 2. | Definitions | 3 |
| | 2.1. Grid Lines | |
| | 2.1.1. Vertical | |
| | 2.1.2. Horizontal | |
| | 2.2. Objects | |
| | 2.2.1. Plates | 3 |
| | 2.3. Properties | 3 |
| | 2.3.1 Concrete | |
| | 2.3.2. Reinforcement | |
| | 2.3.3. Plate Cracking Coefficients | 4 |
| | 2.3.4. Plate Design Criteria | |
| | 2.4. Restraints | |
| | 2.4.1. Supports | |
| | 2.5. Load Case/Combo. | |
| | 2.5.1. Load Cases | |
| | 2.5.2. Load Combinations | |
| 3 | Assignments | |
| ٠. | 3.1. Nodes | |
| | 3.2. Plates | |
| | 3.3. Stiffeners | |
| | 3.4. Point Loads | |
| | 3.5. Uniform Area Loads | |
| 1 | Results | |
| τ. | 4.1. Service | |
| | 4.1.1. Nodal Displacements | |
| | 4.1.1.1.1.2D+1.6Lr+0.5W | |
| | 4.2. Ultimate | |
| | 4.2.1. Nodal Displacements | |
| | 4.2.1.1 1.2D+1.6Lr+0.5W | |
| | 4.2.2. Wall Cross-Sectional Forces | |
| | 4.2.2.1. 1.2D+1.6Lr+0.5W | |
| = | Screenshots | |
| ٥. | 5.1. Extrude 3D view | |
| | 5.1. Extrude 3D view | |
| | 5.3. Nodes ID | |
| | | |
| | 5.4. Restraints | |
| | 5.6. Loads - Case C - W | |
| | | 15 |
| | 0 / 1080S - CaSe D - H | 1.7 |





Page | 3 1/27/2023 4:11 PM

1. Project

1.1. General Information

| File Name | Tilt-Up Wall Panel.walx |
|-----------|---|
| Project | Reinforced Concrete Tilt-Up Wall Panel |
| Code | ACI 318-14 |
| Units | English |
| Date | 11/7/2022 |
| Time | 10:30 PM |

1.2. Solver Options

| Include 2nd order effects | Yes |
|---|----------|
| Check out-of-plane service deflections | Yes |
| Maximum permissible out-of-plane deflections | 2.480 in |
| Check concrete shear strength of wall crossection | No |

2. Definitions

2.1. Grid Lines

2.1.1. Vertical

| Spacing | Coordinate-X | Label | |
|---------|--------------|-------|--|
| ft | ft | | |
| 0.00 | 0.00 | Α | |
| 15.00 | 15.00 | В | |

2.1.2. Horizontal

| Spacing | Coordinate-Y | Label |
|---------|--------------|-------|
| ft | ft | |
| 0.00 | 0.00 | 0 |
| 29.50 | 29.50 | 1 |
| 1.50 | 31.00 | 2 |

2.2. Objects

2.2.1. Plates

| Label | Thickness in | Concrete | Reinforcement | Design Criteria | Cracking Coeff. | Used |
|-------|--------------|----------|---------------|-----------------|-----------------|------|
| W6 | 6.25 | C4 | Gr60 | One layer | PCC1 | Yes |

2.3. Properties

2.3.1. Concrete

| Label | f'c | W _c | Ec | ٧ | Precast | Used |
|-------|--------|----------------|--------|------|---------|------|
| | ksi | pcf | ksi | - | | |
| C4 | 4.0000 | 150.00 | 3605.0 | 0.20 | No | Yes |

2.3.2. Reinforcement

| Label | f _y | E _s | Used | Label | f _y | Es | Used |
|-------|----------------|----------------|------|-------|----------------|-----|------|
| | ksi | ksi | | | ksi | ksi | |
| Gr60 | 60.0000 | 29000.0 | Yes | | | | |





Page | 4 1/27/2023 4:11 PM

2.3.3. Plate Cracking Coefficients

| Label | Service Cor | mbinations | Ultimate Combinations | | Used |
|-------|-------------|--------------|-----------------------|--------------|------|
| | In-plane | Out-of-plane | In-plane | Out-of-plane | |
| PCC1 | 1 | 1 | 1 | 0.07241 | Yes |

2.3.4. Plate Design Criteria

NOTE: Bar centroid location measured from Z-ve face for Back Curtain and Z+ve face for Front Curtain

| | Reinforcement Ratio | | | Reinforcement Location | | | | | | | |
|-----------|---------------------|-------|---------------|------------------------|---------------|---------------|-----------------|-----------------|------------------|------------------|------|
| Label | Curtains | Flags | Rmin (Hor) | Rmax (Hor) | Rmin (Ver) | Rmax (Ver) | Back H. (BH) | Back V. (BV) | Front H. (FH) | Front V. (FV) | Used |
| | | | % | % | % | % | in | in | in | in | |
| One laver | 1 | | 0.20 | 8.00 | 0.15 | 8.00 | 3.13 | 3.13 | | | Yes |

2.4. Restraints

2.4.1. Supports

| | Tra | nslations | | Rotations | | | |
|---------|-------|-----------|-------|-----------|------|------|------|
| Label | Dx | Dy | Dz | Rx | Ry | Rz | Used |
| Pinned | Fixed | Fixed | Fixed | Free | Free | Free | Yes |
| Lateral | Free | Free | Fixed | Free | Free | Free | Yes |

2.5. Load Case/Combo.

2.5.1. Load Cases

NOTE: Self weight is included under Case A.

| Case | Туре | Case Label | Load Defined? |
|------|--------|------------|---------------|
| A | Dead | DL | Yes |
| В | Live | LL | No |
| С | Wind | W | Yes |
| D | Others | Lr | Yes |

2.5.2. Load Combinations

| Combo./Case | Α | В | С | D | E | F | G | н | 1 | Combo Type |
|--------------|-------|-------|-------|--------|---|---|---------------|---|---|---------------|
| Туре | Dead | Live | Wind | Others | | | | | | 120 |
| Combo./Label | DL | LL | W | Lr | | | | | | |
| 1.2D+1.6L | 1.000 | 0.500 | 0.438 | 0.000 | - | - | - | - | - | Ser. |
| 1.2D+1.6L | 1.200 | 0.000 | 0.500 | 1.600 | _ | 2 | (C <u>a</u>) | _ | 2 | Ult. |

3. Assignments

3.1. Nodes

| ID | X Coord. | Y Coord. | Rigid Support | Spring Support |
|----|----------|----------|---------------|----------------|
| | ft | ft | | |
| N1 | 3.00 | 29.50 | | |
| N2 | 7.50 | 29.50 | | |
| N3 | 12.00 | 29.50 | | |

3.2. Plates

| ID | Label | Shape | Top Left/Center X | Top Left/Center Y | Width (B) | Height (H)/Dia. (D) |
|----|-------|-----------|-------------------|-------------------|-----------|---------------------|
| | | | ft | ft | ft | ft |
| P1 | W6 | Polygonal | 7.50 | 15.50 | 15.00 | 31.00 |





Page | 5 1/27/2023 4:11 PM

3.3. Stiffeners

| ID | Label | Direction | Start X | End X | Start Y | End Y | Length | Rigid Support |
|----|----------|------------|---------|-------|---------|-------|--------|------------------|
| | | | ft | ft | ft | ft | ft | |
| S1 | - Null - | Horizontal | 0.00 | 15.00 | 0.00 | 0.00 | 15.00 | Pinned |
| S3 | - Null - | Horizontal | 0.00 | 15.00 | 29.50 | 29.50 | 15.00 | Lateral |

3.4. Point Loads

| Nodes ID Load Case | | Fx | Fy | Fz | Mx | My | Mz | Ecc |
|-----------------------|------|------|-------|--------|--------|--------|------|------|
| | kips | kips | kips | kip-ft | kip-ft | kip-ft | in | |
| N1 | Α | 0.00 | -2.40 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |
| | D | 0.00 | -2.50 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |
| N2 | Α | 0.00 | -2.40 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |
| | D | 0.00 | -2.50 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |
| N3 | Α | 0.00 | -2.40 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |
| | D | 0.00 | -2.50 | 0.00 | 0.00 | 0.00 | 0.00 | 3.00 |

3.5. Uniform Area Loads

| | Plate ID | Load Case | Wx | Wy | Wz |
|--|----------|-----------|------|------|--------|
| | | | psf | psf | psf |
| | P1 | С | 0.00 | 0.00 | -27.20 |

4. Results

4.1. Service

4.1.1. Nodal Displacements

4.1.1.1. 1.2D+1.6Lr+0.5W

Coordinate System: Global

| | | | | | 1 |
|----------------------------------|------|-------|--------|--------|---|
| 1 | Node | Dx | Dy | Dz | 1 |
| | | in | in | in | J |
| | 239 | 0.000 | -0.001 | -0.251 | |
| | 240 | 0.000 | -0.001 | -0.248 | |
| | 241 | 0.000 | -0.001 | -0.246 | |
| | 242 | 0.000 | -0.001 | -0.244 | |
| | 243 | 0.000 | -0.001 | -0.243 | |
| | 244 | 0.000 | -0.001 | -0.242 | |
| | 245 | 0.000 | -0.001 | -0.241 | |
| at critical | 246 | 0.000 | -0.001 | -0.240 | |
| section $(y = 13.77 \text{ ft})$ | 247 | 0.000 | -0.001 | -0.240 | |
| | 248 | 0.000 | -0.001 | -0.240 | |
| | 249 | 0.000 | -0.001 | -0.241 | |
| | 250 | 0.000 | -0.001 | -0.242 | |
| | 251 | 0.000 | -0.001 | -0.243 | |
| | 252 | 0.000 | -0.001 | -0.244 | |
| | 253 | 0.000 | -0.001 | -0.246 | |
| | 254 | 0.000 | -0.001 | -0.248 | |
| | 255 | 0.000 | -0.001 | -0.251 | |
| | 256 | 0.000 | -0.002 | -0.252 | |
| | 257 | 0.000 | -0.002 | -0.250 | |
| at midheight | 258 | 0.000 | -0.002 | -0.247 | |
| (y = 14.75 ft) | 259 | 0.000 | -0.002 | -0.245 | |
| (y - 14.73 11) | 260 | 0.000 | -0.002 | -0.244 | |
| | 261 | 0.000 | -0.002 | -0.243 | |
| | 262 | 0.000 | -0.002 | -0.242 | |

$$D_{z,avg}=0.244 \ in.$$





Page | 6 1/27/2023 4:11 PM

| | Node | Dx | Dy | Dz |
|-----------------|------|-------|--------|--------|
| | | in | in | in |
| | 263 | 0.000 | -0.002 | -0.242 |
| | 264 | 0.000 | -0.002 | -0.242 |
| at midheight | 265 | 0.000 | -0.002 | -0.242 |
| | 266 | 0.000 | -0.002 | -0.242 |
| | 267 | 0.000 | -0.002 | -0.243 |
| (y = 14.75 ft) | 268 | 0.000 | -0.002 | -0.244 |
| | 269 | 0.000 | -0.002 | -0.245 |
| | 270 | 0.000 | -0.002 | -0.247 |
| | 271 | 0.000 | -0.002 | -0.250 |
| | 272 | 0.000 | -0.002 | -0.252 |

 $D_{z,avg} = 0.245 \ \text{in.}$

4.2. Ultimate

4.2.1. Nodal Displacements

4.2.1.1. 1.2D+1.6Lr+0.5W

Coordinate System: Global

| | | 30.00 (1.00 1 .00 (1.00) | ciii. Giobai | | | | |
|------------|--------|--------------------------------------|--------------|--------|--------|--------|--|
| | 1 | lode | Dx | Dy | ſ | Dz | |
| | | | in | in | | in | |
| | | 239 | 0.000 | -0.002 | | -9.890 | |
| | | 240 | 0.000 | -0.002 | | -9.778 | |
| | | 241 | 0.000 | -0.002 | | -9.686 | |
| | | 242 | 0.000 | -0.002 | | -9.612 | |
| | | 243 | 0.000 | -0.002 | | -9.560 | |
| | | 244 | 0.000 | -0.002 | | -9.520 | |
| | | 245 | 0.000 | -0.002 | | -9.492 | |
| at cri | itical | 246 | 0.000 | -0.002 | | -9.476 | |
| sec | ction | 247 | 0.000 | -0.002 | | -9.470 | |
| (y = 13.7) | 77 ft) | 248 | 0.000 | -0.002 | | -9.476 | |
| 0 | / | 249 | 0.000 | -0.002 | | -9.492 | |
| | 250 | 0.000 | -0.002 | | -9.520 | | |
| | | 251 | 0.000 | -0.002 | | -9.560 | |
| | 252 | 0.000 | -0.002 | | -9.612 | | |
| | 253 | 0.000 | -0.002 | | -9.686 | | |
| | 254 | 0.000 | -0.002 | | -9.778 | | |
| | | 255 | 0.000 | -0.002 | | -9.890 | |
| | | 256 | 0.000 | -0.002 | | -9.919 | |
| | | 257 | 0.000 | -0.002 | | -9.807 | |
| | | 258 | 0.000 | -0.002 | | -9.715 | |
| | | 259 | 0.000 | -0.002 | | -9.642 | |
| | | 260 | 0.000 | -0.002 | | -9.589 | |
| | | 261 | 0.000 | -0.002 | | -9.550 | |
| | | 262 | 0.000 | -0.002 | | -9.522 | |
| at midh | eight | 263 | 0.000 | -0.002 | | -9.505 | |
| (y = 14.7) | | 264 | 0.000 | -0.002 | | -9.500 | |
| (y = 14. | 13 It) | 265 | 0.000 | -0.002 | | -9.505 | |
| | | 266 | 0.000 | -0.002 | | -9.522 | |
| | | 267 | 0.000 | -0.002 | | -9.550 | |
| | | 268 | 0.000 | -0.002 | | -9.589 | |
| | | 269 | 0.000 | -0.002 | | -9.642 | |
| | | 270 | 0.000 | -0.002 | | -9.715 | |
| | | 271 | 0.000 | -0.002 | | -9.807 | |
| | | 272 | 0.000 | -0.002 | | -9.919 | |

 $D_{z,avg} = 9.618 \ in. \label{eq:decomposition}$

 $D_{z,avg} = 9.647$ in.





Page | **7** 1/27/2023 4:11 PM

4.2.2. Wall Cross-Sectional Forces 4.2.2.1. 1.2D+1.6Lr+0.5W

24-

24+

22.62

22.62

7.50

7.50

0.00

0.00

-32.43

-32.43

0.00

0.00

4.19

4.19

39.30

39.30

0.00

0.00

Coordinate System: Global (+) Horizontal cross-section above Y-coordinate

| | Wall Crossection | In-Plane Forces | | | Out-Of-Plane Forces | | | |
|--------------|------------------|-----------------|------|--------|---------------------|-------|--------|------|
| No. | Y coordinate | X-Centroid | Vux | Nuy | Muz | Vuz | Mux | Mu |
| | ft | ft | kips | kips | kip-ft | kips | kip-ft | kip- |
| 1- | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.0 |
| 1+ | 0.00 | 7.50 | 0.00 | -64.23 | 0.00 | -8.15 | 0.00 | 0.0 |
| 2- | 0.98 | 7.50 | 0.00 | -62.85 | 0.00 | -7.79 | 7.84 | 0.0 |
| 2+ | 0.98 | 7.50 | 0.00 | -62.85 | 0.00 | -7.79 | 7.84 | 0.0 |
| 3- | 1.97 | 7.50 | 0.00 | -61.47 | 0.00 | -7.36 | 15.29 | 0.0 |
| 3+ | 1.97 | 7.50 | 0.00 | -61.47 | 0.00 | -7.36 | 15.29 | 0.0 |
| 4- | 2.95 | 7.50 | 0.00 | -60.09 | 0.00 | -6.88 | 22.30 | 0.0 |
| 4+ | 2.95 | 7.50 | 0.00 | -60.09 | 0.00 | -6.88 | 22.29 | 0.0 |
| 5- | 3.93 | 7.50 | 0.00 | -58.70 | 0.00 | -6.34 | 28.80 | 0.0 |
| 5+ | 3.93 | 7.50 | 0.00 | -58.70 | 0.00 | -6.34 | 28.80 | 0.0 |
| 6- | 4.92 | 7.50 | 0.00 | -57.32 | 0.00 | -5.76 | 34.75 | 0.0 |
| 6+ | 4.92 | 7.50 | 0.00 | -57.32 | 0.00 | -5.76 | 34.75 | 0.0 |
| 7- | 5.90 | 7.50 | 0.00 | -55.94 | 0.00 | -5.15 | 40.12 | 0.0 |
| 7+ | 5.90 | 7.50 | 0.00 | -55.94 | 0.00 | -5.15 | 40.12 | 0.0 |
| 8- | 6.88 | 7.50 | 0.00 | -54.55 | 0.00 | -4.51 | 44.87 | 0.0 |
| 8+ | 6.88 | 7.50 | 0.00 | -54.55 | 0.00 | -4.51 | 44.87 | 0.0 |
| 9- | 7.87 | 7.50 | 0.00 | -53.17 | 0.00 | -3.86 | 48.99 | 0.0 |
| 9+ | 7.87 | 7.50 | 0.00 | -53.17 | 0.00 | -3.86 | 48.99 | 0.0 |
| 10- | 8.85 | 7.50 | 0.00 | -51.79 | 0.00 | -3.20 | 52.46 | 0.0 |
| 10+ | 8.85 | 7.50 | 0.00 | -51.79 | 0.00 | -3.20 | 52.46 | 0.0 |
| 11- | 9.83 | 7.50 | 0.00 | -50.41 | 0.00 | -2.53 | 55.28 | 0.0 |
| 11+ | 9.83 | 7.50 | 0.00 | -50.41 | 0.00 | -2.53 | 55.27 | 0.0 |
| 12- | 10.82 | 7.50 | 0.00 | -49.02 | 0.00 | -1.87 | 57.44 | 0.0 |
| 12+ | 10.82 | 7.50 | 0.00 | -49.02 | 0.00 | -1.87 | 57.43 | 0.0 |
| 13- | 11.80 | 7.50 | 0.00 | -47.64 | 0.00 | -1.21 | 58.95 | 0.0 |
| 13+ | 11.80 | 7.50 | 0.00 | -47.64 | 0.00 | -1.21 | 58.95 | 0.0 |
| 14- | 12.78 | 7.50 | 0.00 | -46.26 | 0.00 | -0.58 | 59.83 | 0.0 |
| 14+ | 12.78 | 7.50 | 0.00 | -46.26 | 0.00 | -0.58 | 59.83 | 0.0 |
| critical 15- | 13.77 | 7.50 | 0.00 | -44.87 | 0.00 | 0.04 | 60.09 | 0.0 |
| section 15+ | 13.77 | 7.50 | 0.00 | -44.87 | 0.00 | 0.04 | 60.09 | 0.0 |
| dheight 16- | 14.75 | 7.50 | 0.00 | -43.49 | 0.00 | 0.63 | 59.76 | 0.0 |
| dheight 16+ | 14.75 | 7.50 | 0.00 | -43.49 | 0.00 | 0.63 | 59.75 | 0.0 |
| 17- | 15.73 | 7.50 | 0.00 | -42.11 | 0.00 | 1.20 | 58.85 | 0.0 |
| 17+ | 15.73 | 7.50 | 0.00 | -42.11 | 0.00 | 1.20 | 58.85 | 0.0 |
| 18- | 16.72 | 7.50 | 0.00 | -40.73 | 0.00 | 1.73 | 57.41 | 0.0 |
| 18+ | 16.72 | 7.50 | 0.00 | -40.73 | 0.00 | 1.73 | 57.41 | 0.0 |
| 19- | 17.70 | 7.50 | 0.00 | -39.34 | 0.00 | 2.23 | 55.46 | 0.0 |
| 19+ | 17.70 | 7.50 | 0.00 | -39.34 | 0.00 | 2.23 | 55.46 | 0.0 |
| 20- | 18.68 | 7.50 | 0.00 | -37.96 | 0.00 | 2.69 | 53.04 | 0.0 |
| 20+ | 18.68 | 7.50 | 0.00 | -37.96 | 0.00 | 2.69 | 53.03 | 0.0 |
| 21- | 19.67 | 7.50 | 0.00 | -36.58 | 0.00 | 3.12 | 50.17 | 0.0 |
| 21+ | 19.67 | 7.50 | 0.00 | -36.58 | 0.00 | 3.12 | 50.17 | 0.0 |
| 22- | 20.65 | 7.50 | 0.00 | -35.19 | 0.00 | 3.52 | 46.90 | 0.0 |
| 22+ | 20.65 | 7.50 | 0.00 | -35.19 | 0.00 | 3.52 | 46.90 | 0.0 |
| 23- | 21.63 | 7.50 | 0.00 | -33.81 | 0.00 | 3.87 | 43.27 | 0.0 |
| 23+ | 21.63 | 7.50 | 0.00 | -33.81 | 0.00 | 3.87 | 43.26 | 0.0 |
| | | | | | | | | |





Page | 8 1/27/2023 4:11 PM

| | Wall Crossection | | In-P | lane Forces | | Out-Of | -Plane Forces | |
|-----|------------------|------------|------|-------------|--------|--------|---------------|--------|
| No. | Y coordinate | X-Centroid | Vux | Nuy | Muz | Vuz | Mux | Muy |
| | ft | ft | kips | kips | kip-ft | kips | kip-ft | kip-ft |
| 25- | 23.60 | 7.50 | 0.00 | -31.05 | 0.00 | 4.47 | 35.04 | 0.00 |
| 25+ | 23.60 | 7.50 | 0.00 | -31.05 | 0.00 | 4.47 | 35.04 | 0.00 |
| 26- | 24.58 | 7.50 | 0.00 | -29.66 | 0.00 | 4.72 | 30.52 | 0.00 |
| 26+ | 24.58 | 7.50 | 0.00 | -29.66 | 0.00 | 4.72 | 30.51 | 0.00 |
| 27- | 25.57 | 7.50 | 0.00 | -28.28 | 0.00 | 4.94 | 25.77 | 0.00 |
| 27+ | 25.57 | 7.50 | 0.00 | -28.28 | 0.00 | 4.94 | 25.76 | 0.00 |
| 28- | 26.55 | 7.50 | 0.00 | -26.90 | 0.00 | 5.12 | 20.82 | 0.00 |
| 28+ | 26.55 | 7.50 | 0.00 | -26.90 | 0.00 | 5.12 | 20.82 | 0.00 |
| 29- | 27.53 | 7.50 | 0.00 | -25.52 | 0.00 | 5.28 | 15.70 | 0.00 |
| 29+ | 27.53 | 7.50 | 0.00 | -25.52 | 0.00 | 5.28 | 15.70 | 0.00 |
| 30- | 28.52 | 7.50 | 0.00 | -24.13 | 0.00 | 5.41 | 10.44 | 0.00 |
| 30+ | 28.52 | 7.50 | 0.00 | -24.13 | 0.00 | 5.41 | 10.44 | 0.00 |
| 31- | 29.50 | 7.50 | 0.00 | -22.75 | 0.00 | 5.52 | 5.07 | 0.00 |
| 31+ | 29.50 | 7.50 | 0.00 | -2.11 | 0.00 | -0.12 | -0.09 | 0.00 |
| 32- | 30.25 | 7.50 | 0.00 | -1.05 | 0.00 | -0.06 | -0.02 | 0.00 |
| 32+ | 30.25 | 7.50 | 0.00 | -1.05 | 0.00 | -0.06 | -0.02 | 0.00 |
| 33- | 31.00 | 7.50 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |
| 33+ | 31.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 | 0.00 |

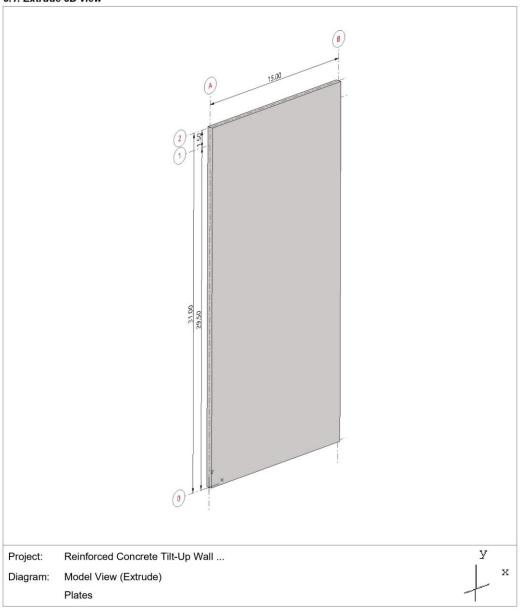




Page | 9 1/27/2023 4:11 PM

5. Screenshots

5.1. Extrude 3D view

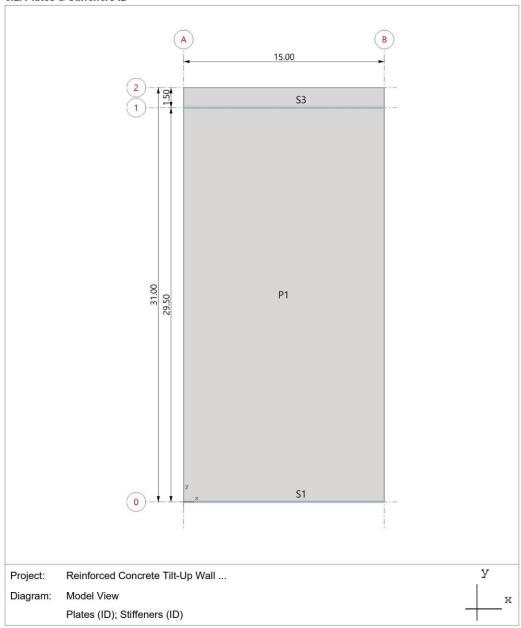






Page | 10 1/27/2023 4:11 PM

5.2. Plates & Stiffeners ID

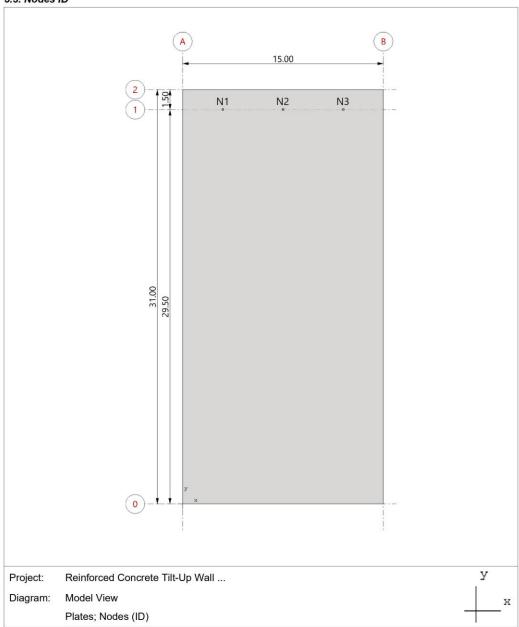






Page | 11 1/27/2023 4:11 PM

5.3. Nodes ID

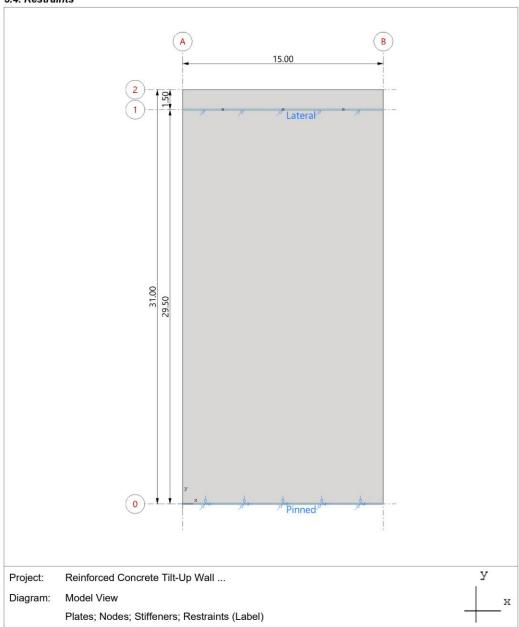






Page | **12** 1/27/2023 4:11 PM

5.4. Restraints





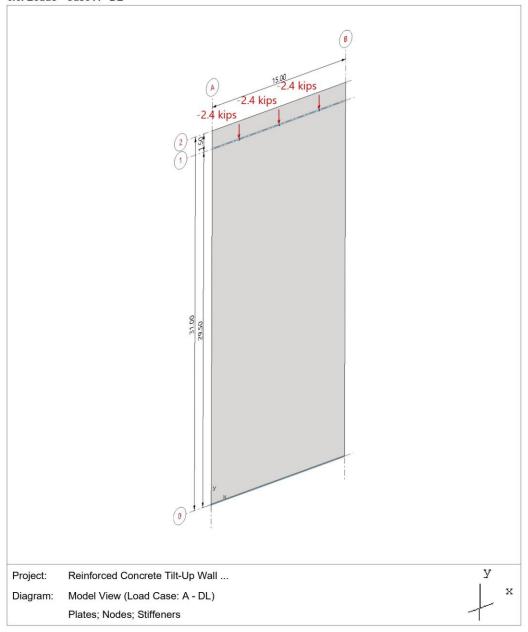


→ Wai

STRUCTUREPOINT - spWall v10.00 (TM)
Licensed to: StructurePoint, LLC, License ID: 00000-0000000-4-20FC1-20FC1
E:\StructurePoint\spWall\Tilt-Up Wall Panel.walx

Page | 13 1/27/2023 4:11 PM

5.5. Loads - Case A - DL

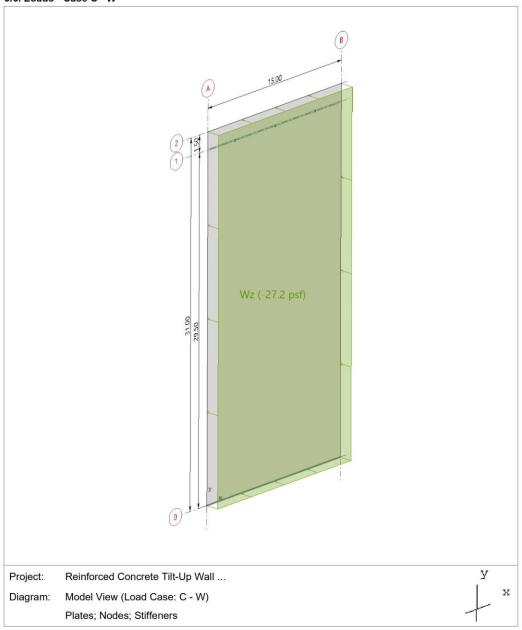






Page | 14 1/27/2023 4:11 PM

5.6. Loads - Case C - W

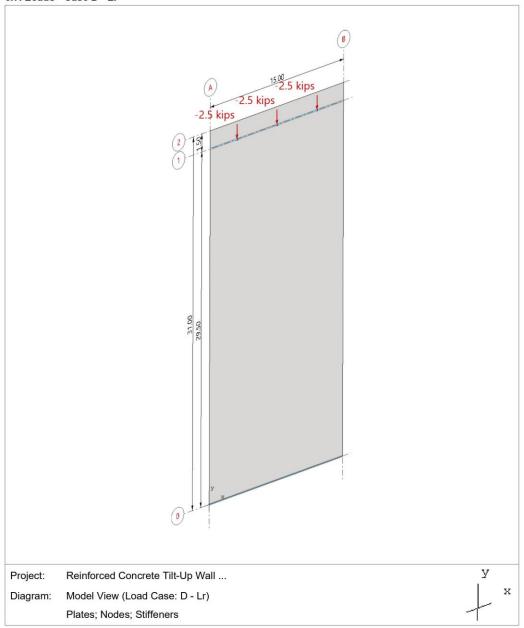






Page | **15** 1/27/2023 4:11 PM

5.7. Loads - Case D - Lr







10. Design Results Comparison and Conclusions

| Table 1 – Comparison of Tilt-Up Wall Panel Analysis and Design Results | | | | | | | |
|--|-------------------------|-----------------------|------------------------------|-------------------------------|--|--|--|
| Solution | M _u (kip-ft) | N _u (kips) | D _{z,service} (in.) | D _{z,ultimate} (in.) | | | |
| Hand (at midheight) | 61.00 | 43.49 | 0.247 | 9.995 | | | |
| spWall (at midheight)* | 59.76 | 43.49 | 0.245 | 9.647 | | | |
| spWall (at critical section)** | 60.09 | 44.87 | 0.244 | 9.618 | | | |

Values are taken at midheight (y = 14.75 ft) for comparison purposes with hand calculations.

The results of all the hand calculations illustrated above are in agreement with the automated exact results obtained from the <u>spWall</u> program.

In column and wall analysis, section properties shall be determined by taking into account the influence of axial loads, the presence of cracked regions along the length of the member, and the effect of load duration (creep effects). ACI 318 permits the use of moment of inertia values of 0.70 I_g for uncracked walls and 0.35I_g for cracked walls.

ACI 318-14 (6.6.3.1.1)

In <u>spWall</u> program, these effects are accounted for where the user can input reduced moment of inertia using "cracking coefficient" values for plate and stiffener elements to effectively reduce stiffness. Cracking coefficients for out-of-plane (bending and torsion) and in-plane (axial and shear) stiffness can be entered for plate elements. Because the values of the cracking coefficients can have a large effect on the analysis and design results, the user must take care in selecting values that best represent the state of cracking at the particular loading stage. Cracking coefficients are greater than 0 and less than 1.

At ultimate loads, a wall is normally in a highly cracked state. The user could enter a value of out-of-plane cracking coefficient for plates of $I_{cracked}/I_{gross}$ based on estimated values of A_s after the analysis and design, if the computed value of A_s greatly differs from the estimated value of A_s , the analysis should be performed again with new values for the cracking coefficients. To account for variations in material properties and workmanship, a factor of 0.75 can be used to reduce the calculated bending stiffness of the concrete section in accordance with ACI 318-14 Chapter 11. ACI 551 states, this bending stiffness reduction factor should be incorporated by all other design methods to comply with the requirements of ACI 318.

At service loads, a wall may or may not be in a highly cracked state. For service load deflection analysis, a wall panel should be modeled with an out-of-plane cracking coefficient of ($I_{effective}/I_{gross}$).

Based on the previous discussion, the ratio between I_{cr} and I_g including a reduction factor of 0.75 is used as the cracking coefficient for the out-of-plane case for ultimate load combinations. In this example, I_{cr} and I_g were found to be equal

^{**} Values are taken at critical section (y = 13.77 ft) with maximum moment value.





to 353.56 in.⁴ and 3,662 in.⁴. Thus, the out-of-plane cracking coefficient for ultimate load combinations can be found as follows:

$$\alpha = \text{cracking coefficient} = \frac{0.75 \times I_{cr}}{I_g} = \frac{0.75 \times 353.56}{3,662} = 0.07241$$

For the service load combinations, M_a equals 20.84 ft-kip which is less than $M_{cr} = 46.32$ ft-kip indicating the section is uncracked ($I_{effective} = I_{gross}$) and the cracking coefficient can be set to 1.0

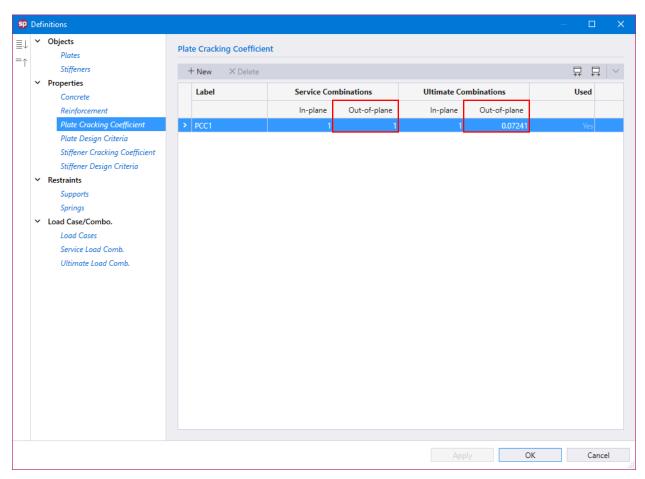


Figure 12 – Defining Cracking Coefficient (spWall)